

# Identification of flutter derivatives from full-scale ambient vibration measurements of the Clifton Suspension Bridge

Nikolaos Nikitas<sup>\*1</sup>, John H.G. Macdonald<sup>1</sup> and Jasna B. Jakobsen<sup>2</sup>

<sup>1</sup>*Department of Civil Engineering  
University of Bristol, University Walk, Bristol, BS8 1TR, UK*

<sup>2</sup>*Department of Mechanical and Structural Engineering and Material Science  
University of Stavanger, N-4036, Stavanger, Norway*

*(Received June 21, 2010, Accepted December 1, 2010)*

**Abstract.** The estimated response of large-scale engineering structures to severe wind loads is prone to modelling uncertainties that can only ultimately be assessed by full-scale testing. To this end ambient vibration data from full-scale monitoring of the historic Clifton Suspension Bridge has been analysed using a combination of a frequency domain system identification method and a more elaborate stochastic identification technique. There is evidence of incipient coupling action between the first vertical and torsional modes in strong winds, providing unique full-scale data and making this an interesting case study. Flutter derivative estimation, which has rarely previously been attempted on full-scale data, was performed to provide deeper insight into the bridge aerodynamic behaviour, identifying trends towards flutter at higher wind speeds. It is shown that, as for other early suspension bridges with bluff cross-sections, single-degree-of-freedom flutter could potentially occur at wind speeds somewhat below requirements for modern designs. The analysis also demonstrates the viability of system identification techniques for extracting valuable results from full-scale data.

**Keywords:** full-scale; system identification; ambient vibration data; flutter derivatives.

---

## 1. Introduction

For full-scale structures the most rational way to proceed with predictions of the reliability and operational safety includes identification methods from response only measurements. Especially for existing bridges, treatment of the flutter instability can substantially be verified in this way. No analytical solutions exist for the fluid forces on the bluff bridge cross-sections to identify the critical wind speed, so inevitably each investigation has to adopt some experimental or semi-empirical foundation to proceed to a further assessment. Most commonly wind tunnel tests of scale models are used for reproducing the flutter phenomenon leaving the question of the effects of scaling issues. Even minor details such as deck railings or roadway grills and vents can strongly alter aerodynamic performance (see Jones *et al.* 1995 and Matsumoto *et al.* 2001). Hence, only analysis of the response of the real bridge can clarify the validity of wind tunnel tests and even reveal aspects, which either due to modelling assumptions or to loading irregularities, were previously concealed.

---

\* Corresponding Author, Dr., E-mail: [N.Nikitas@bristol.ac.uk](mailto:N.Nikitas@bristol.ac.uk)

Aeroelastic parameters have rarely been obtained from full-scale bridge data. Jakobsen and Larose (1999) addressed the problem on the Høga Kusten Bridge and presented a comparative analysis with wind tunnel results using a subspace identification technique for extraction of flutter derivatives. Costa and Borri (2007) essentially applied the same identification routine (Jakobsen 1995, Jakobsen and Larose 1999), both on numerically simulated responses and on measured data from the Iroise Bridge, finding good performance of the method in each case. For both of these bridge studies, the identification routine itself was found to be reliable when tested using simulated data (including with added noise) or against other methods for wind tunnel data. Comparisons between full-scale and wind tunnel results were not unreasonable, but since the full-scale bridges were far from flutter, the trends in the flutter derivatives were not very clear.

Another approach to the problem of identifying the aerodynamic effects on full-scale bridge vibration characteristics, but without quantifying flutter derivatives, was used by Macdonald (2003) on the Second Severn Crossing. Variations of effective damping ratios and natural frequencies with wind speed were found and some indications of aeroelastic modal coupling were identified on the partially constructed bridge (Macdonald and Daniell 2005). In other full-scale studies, Littler (1992) and Brownjohn (1994) on the Humber Bridge, Bietry *et al.* (1994) on the Saint-Nazaire Bridge, Jensen *et al.* (1999) on the Great Belt Bridge, Ge and Tanaka (2002) on the Høga Kusten Bridge during construction and Nagayama *et al.* (2005) on the Hakucho Bridge all found some trends of effective aerodynamic damping with wind speed, but coupling between modes and flutter derivatives were not identified.

The limited number of full-scale studies from which aeroelastic parameters have been found makes any new cases useful for furthering knowledge of the viability of such system identification from site data and for interpreting actual bridge behaviour.

## 2. Case study

In the current paper, analysis is performed on full-scale vibration measurements from the historic Clifton Suspension Bridge (CSB), shown in Fig. 1. The CSB spans the Avon Gorge in Bristol, UK and was designed by I.K. Brunel in 1830, although it was not completed until 1864 (Barlow 1867). It was one of the longest suspension bridges of that time, with a main span of 214 m. Wrought iron chains provide the suspension system, being the common practice for such early long-span bridges. In the light of modern understanding of bridge aerodynamics, the bridge cross-section (Fig. 2) and its light weight make it potentially susceptible to wind-induced vibrations. Indeed, on a few occasions in its lifespan large amplitude vibrations in strong winds have been reported.

On Christmas Day 1990 there was evidence of vertical motion at the bridge ends of the order of 250 mm, which translates to even larger amplitudes within the bridge span. Both vertical and torsional deck motions were evident on a video recording of the bridge towards the end of the storm. A similar large vibration event was reported on 3 December 2006. Although no wind recordings exist from the bridge site itself on these occasions, data from the nearest weather stations imply that the maximum 1-hour mean wind speeds could have been around 20 m/s. For recordings on site with wind speeds up to 16 m/s, coupling action between the first vertical and torsional modes seemed to occur (Section 4.2) and the maximum vertical displacement of the end of the bridge was 35 mm (and the maximum measured elsewhere 92 mm). The coupling action between modes and the rapid growth of vibration amplitudes for a modest increase in wind speed indicate

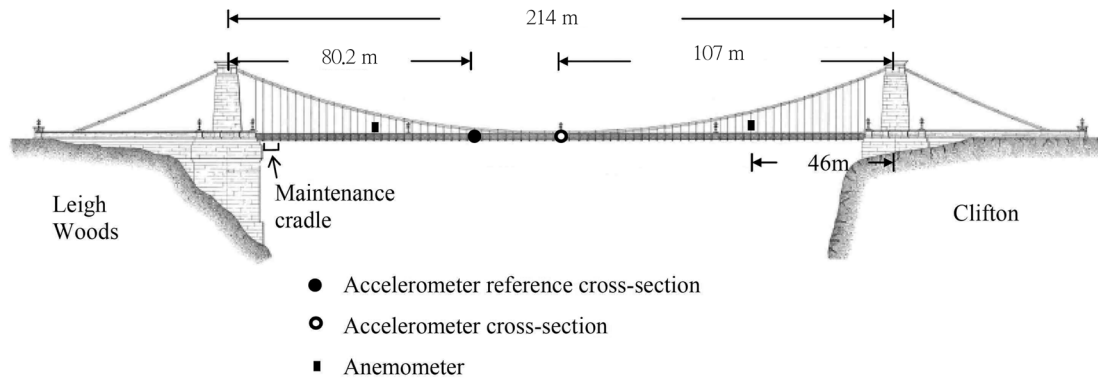


Fig. 1 Bridge elevation showing instrument locations. Based on figure after Barlow (1867), with permission from Thomas Telford Publishing

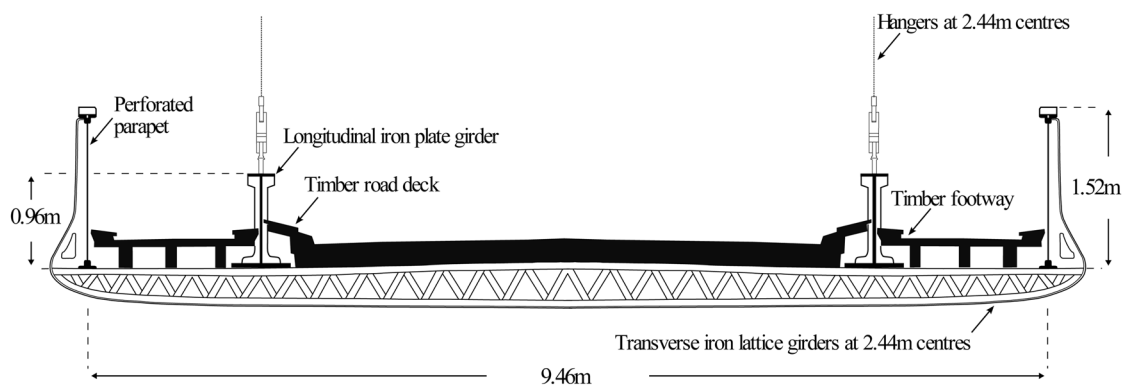


Fig. 2 Sketch of the bridge cross-section

strong aeroelastic effects, making the bridge behaviour rather interesting. Such characteristics are reminiscent of features observed, in a more severe form, on the Tacoma Narrows (Farquharson *et al.* 1950-1954) and Deer Isle (Kumarasena *et al.* 1989a,b) bridges.

It is worth noting that ten suspension bridges from the same era as the CSB failed due to wind between 1818 and 1889, including the Menai Straits Bridge and a span of the Brighton Chain Pier (Farquharson *et al.* 1950-1954). In contrast the CSB has survived virtually unscathed for over 140 years. According to empirical estimates, as in most bridge design rules, it is potentially susceptible to flutter with an estimated critical wind speed of only around 20 m/s. Therefore adverse aeroelastic effects could become significant in moderately strong winds.

For the current study, the wind conditions and bridge response recordings over two winter periods, from November 2003 to March 2004 and from December 2007 to February 2008, are used. The data include several occasions with moderately strong winds, and reasonable ranges of wind speeds and directions, thus enabling a meaningful assessment of the wind effects on the bridge dynamics.

Two ultrasonic anemometers were mounted 61 m either side of midspan, more than 5 m above road level. Nine accelerometers were positioned at a series of cross-sections along the bridge during

an earlier analysis of the CSB dynamic response, enabling mode shapes to be identified (Macdonald 2008). For the records considered here, two sets of three accelerometers were positioned at midspan and at a cross-section slightly off centre (26.8 m from midspan) as illustrated in Fig. 1. This location was chosen as the reference cross-section since all significant vibration modes could be measured there. Signals from all instruments were passed through anti-aliasing filters with a cut-off frequency of 4 Hz and were recorded at a sampling rate of 12.5 Hz. The primary aim of this study was to uncover details of large amplitude response that the bridge was found to produce for certain wind conditions by determining the variation of modal characteristics with wind velocity and explore the possibility for flutter instability.

Modal parameter estimates from a frequency based curve fitting technique (for details see Macdonald and Daniell 2005) are used here, together with a subspace stochastic identification formulation especially modified to extract flutter derivatives (Jakobsen 1995). In this study, due to lack of wind tunnel data from a scale model, flutter derivatives of other typical bridge cross-sections, as presented by Scanlan and Tomko (1971), are used to assess potential similarities. Cross-sections employed for comparative purposes are chosen to represent both the low structural depth and the high parapets (perforated on the CSB), of the section in hand.

The paper consists first of a short discussion on the acquired wind measurements. Typical wind speeds and wind turbulence conditions are described and the local terrain effects are discussed. Subsequently attention is moved to the bridge response details. A conventional modal analysis is performed based on linear modelling and all modal characteristics are identified. The last part containing the flutter derivative identification scheme first presents the employed Covariance Block Hankel Matrix (CBHM) formulation and subsequently discusses the flutter potential of the CSB case study.

### **3. Wind characteristics**

The topography around the CSB has a considerable effect on the local wind characteristics. As shown in the polar plots from both anemometers in Fig. 3, the wind speeds follow certain trends with orientations. (True North is at  $31^\circ$  relative to the axes shown). The trends differ markedly from the general wind pattern in the region away from the local effects. The strongest winds in the absence of topographic effects are typically from the south-west direction (at about  $250^\circ$  on these axes). The stronger winds, as measured, are aligned along the gorge and can be easily explained by funnelling of the flow in these orientations and sheltering due to the high ground near the bridge ends. Strong winds from the south-west are greatly attenuated at the bridge site, and virtually no wind from the north-east quadrant is experienced at the site. It was also observed that the correlation of the wind directions and wind speeds measured by the two anemometers was strongly influenced by the wind orientation. In particular, for wind directions close to along the gorge the phenomenon was rather pronounced with even a small change in wind direction resulting a large but consistent variation in the ratio between the two individual anemometer wind speed values.

The maximum wind speed, averaged over one hour (unless stated all wind information hereafter refers to 1-hour means), did not exceed 16 m/s at the bridge site, although higher speeds were measured at the nearest weather stations, for winds from the south-west. A histogram of 1-hour average wind speeds at the bridge is given in Fig. 3. Maximum 1 second gusts were of the order of 26 m/s for both anemometers.

In addition, the wind turbulence and angle of attack parameters were considered. For wind turbulence there was a strong dependence on wind direction and a weaker one on wind speed. High levels of turbulence (up to 60% longitudinal turbulence intensity) were experienced, particularly for wind not along the gorge and for lower wind speeds. In winds over 8 m/s, which only occurred along the gorge, approximately normal to the bridge, the mean longitudinal turbulence intensity was 21% and the mean vertical turbulence intensity 10%. The vertical and across-wind turbulence intensities followed very similar patterns to the longitudinal turbulence. For longitudinal turbulence intensities up to 40%, the across-wind turbulence was approximately equal to it and the vertical turbulence intensity approximately half of the value. These are typical of relationships between the three components of turbulence. For higher turbulence intensities measured, generally in lower wind speeds, the vertical and across-wind turbulence intensities were relatively larger.

For the vertical angle of attack there was also strong dependence on the wind direction, and there were noticeable differences in the measurements from the two anemometers. The presence of the bridge itself is likely to have affected these measurements, as well as the topography of the gorge, since the anemometers were relatively close to the deck. High vertical angles of attack occurred, up to approximately  $\pm 10^\circ$ . (It should be noted that these values are all averaged over 1 hour periods). There was no significant difference in vertical angles of attack for different wind speeds.

A final wind aspect significant for the subsequent analysis refers to the frequency content of wind

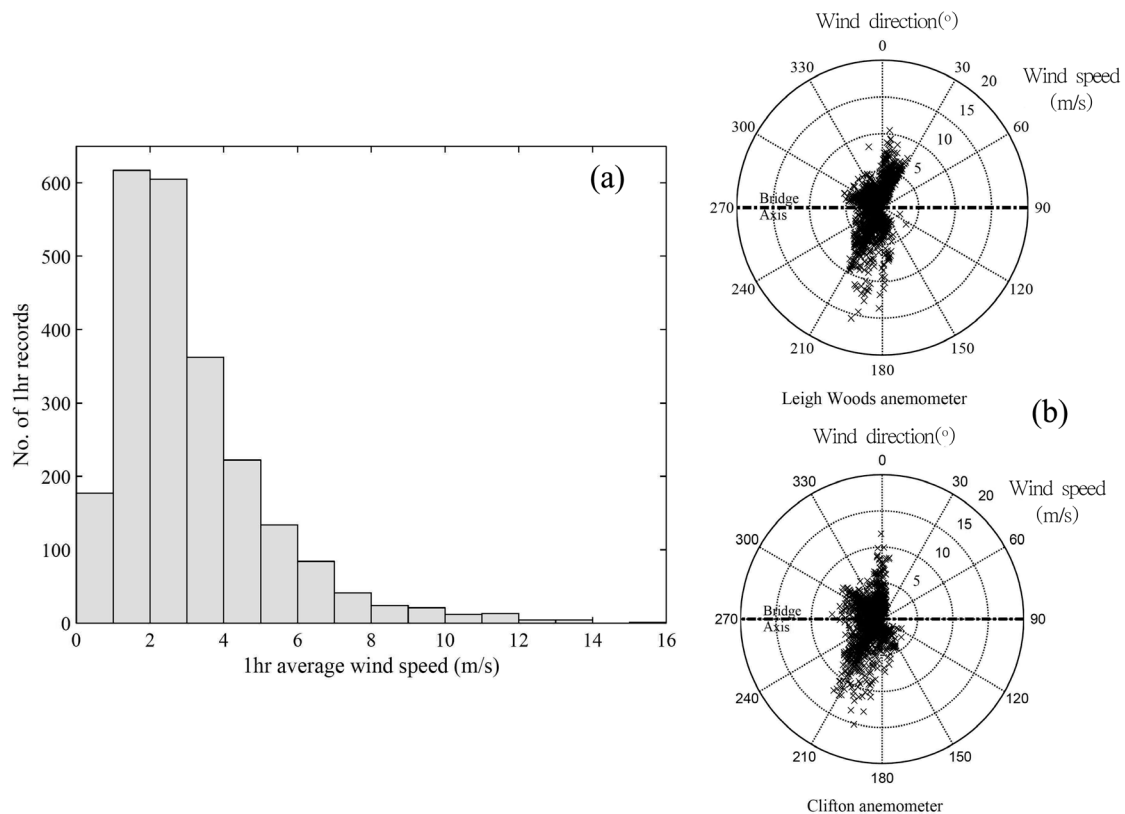


Fig. 3 (a) Histogram of wind speeds during the 2003-04 recording period and (b) polar plots of 1-hour mean wind velocities from both anemometers

buffeting. Although the traffic loading seems to be reasonably well captured by a white noise loading approximation (predominantly from step loading as vehicles drive onto or off the suspended span), the same does not hold for wind. By comparing spectral estimates deduced for various combinations of wind and traffic loading it was found that above approximately 0.25Hz the wind loading spectra appeared as  $1/f$  noise with a power exponent  $-8/3$ , thus producing a general loading spectrum approximated by a relation of the form

$$S_{load}(f) = S_w f^{-8/3} + S_t \quad (1)$$

where:  $S_w$  is a constant for a given record, being a function of the wind parameters,  $f$  is frequency and  $S_t$  is the magnitude of white noise traffic loading for the particular record.

The frequency power exponent of  $-8/3$  is as expected for the ‘high frequency’ tail of the wind buffeting load spectrum, for a 3D aerodynamic admittance function inversely proportional to frequency.

## 4. Response and modal parameters

### 4.1 Response characteristics

Fig. 4 shows the 1-hour average wind speeds over the whole of the first monitoring period, and the corresponding Root Mean Square (RMS) vertical accelerations at the reference cross-section. The RMS amplitudes normally show a clear daily cycle with the varying traffic load, with a maximum vertical response of approximately  $0.02 \text{ m/s}^2$ . By comparison it can be seen that only in wind speeds exceeding approximately  $8 \text{ m/s}$  does the response noticeably exceed the maximum traffic-induced response. The maximum wind-induced acceleration measured was approximately four times the maximum traffic-induced acceleration. The torsional and lateral acceleration responses at the reference cross-section followed very similar patterns to the vertical response over the monitoring period, although the magnitudes of the responses were lower. The maximum instantaneous value of each component was found to be approximately six times the 1-hour RMS value.

Dynamic displacements were calculated from the measured accelerations by double integration and it was noticed that the response is dominated by low frequency modes. The dominance of the low frequency modes is caused by the relatively higher wind loading at low frequencies and the effect of the integration, which exaggerates low frequency components. Whereas the maximum RMS acceleration due to wind loading was approximately four times the maximum due to traffic loading, in terms of displacement the maximum RMS response to wind was approximately 10 times that for traffic.

The influence of wind loading on the measured vertical accelerations is shown in Fig. 5. Similar figures were obtained for the lateral and torsional accelerations. The scatter of results, particularly at low wind speeds, is largely due to the varying traffic contribution. The varying wind turbulence intensity also had an effect. Excluding records dominated by traffic and normalising by the corresponding wind turbulence intensity gives a much clearer relationship with wind speed as shown in Fig. 5(b). Normalised RMS responses are then close to power law functions of the wind speed with a power exponent of approximately 3.

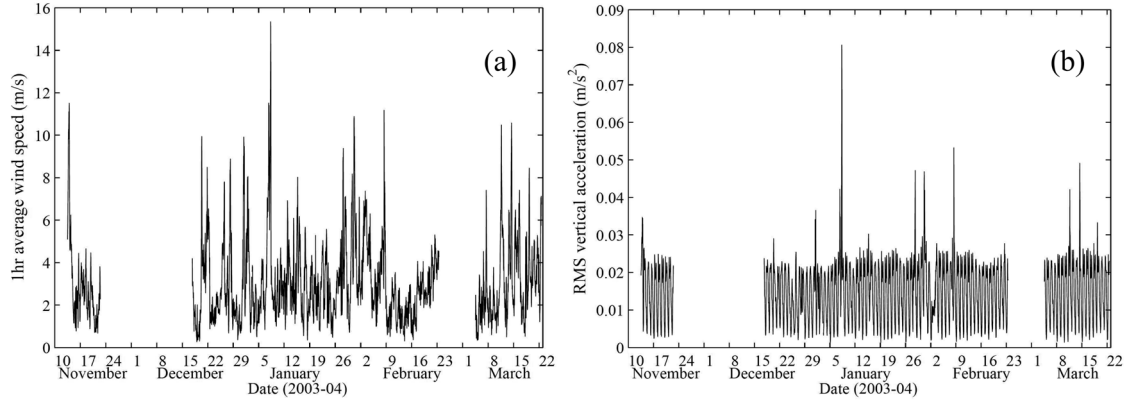


Fig. 4 (a) 1hr average wind speed over the monitoring period and (b) 1-hour RMS vertical accelerations at the reference location over the monitoring period

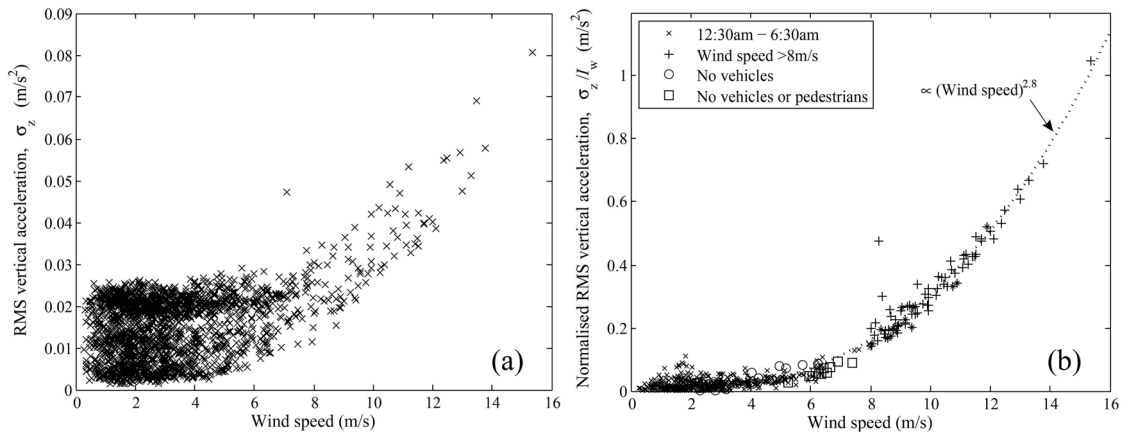


Fig. 5 (a) RMS vertical accelerations in relation to wind speed for all 1hr records and (b) same as (a) for 1hr records dominated by wind loading, with RMS vertical accelerations now divided by the vertical turbulence intensity. The power-law approximating the obtained trend is also plotted

#### 4.2 Modal analysis

Modal parameters were calculated from the acceleration Power Spectral Densities (PSDs) using the Iterative Windowed Curve-fitting Method (Macdonald and Daniell 2005), specifically developed for the analysis of ambient vibration data and allowing for the previously specified loading spectrum. The method iteratively curve fits in the frequency domain a series of idealised single-degree-of-freedom transfer functions, taking into account the loading spectrum, the effect of windowing on the spectra (both from the measured data and equivalently from the idealised transfer functions) and the contributions of multiple modes.

Measurements were only taken on the suspended bridge deck, but all modes inevitably involve

vibrations of other parts of the structure, particularly the chains. Analysis was performed for frequencies up to 3 Hz with twelve vertical, eleven torsional and four lateral modes being identified in this range, based on measurements in low wind speeds (Macdonald 2008). Typical PSDs for three different loading scenarios for vertical, torsional and lateral accelerations are given in Fig. 6 to present the effect of wind loading on the bridge response. An important finding is the proximity of the first vertical and torsional modes, with natural frequencies of 0.293 Hz and 0.356 Hz respectively (ratio 0.82). These are the first antisymmetric modes of each type and it appears that in the stronger winds they start to couple in a potentially incipient flutter motion as evidenced by the small hump at 0.35 Hz in the vertical spectrum, as seen in Fig. 6(a) and more clearly in Fig. 7.

For further analysis it was desirable to reduce the actual multi-mode response to a simpler two-degree-of-freedom problem concerning the first vertical and torsional modes. Low-pass filtering could not fully remove the contributions of the second mode of each type. However, using the measured responses at both accelerometer cross-sections (Fig. 1) together with the previously measured mode shapes (Macdonald 2008), the first of pair modes (vertical and torsional anti-symmetric, hence not measured at midspan) could effectively be isolated from the second pair of modes (vertical and torsional symmetric, measured at midspan), by simple subtraction in the time domain.

Fig. 7(a) hence shows the responses of the first pair of modes for the highest recorded 1-hour wind speed (15.3 m/s). The peak in the vertical PSD at the torsional frequency is strong evidence of coupling action. The coupling was not evident for low winds and became stronger for higher winds (Fig. 7(b)), showing it to be an aeroelastic effect. The coherence between the vertical and torsional

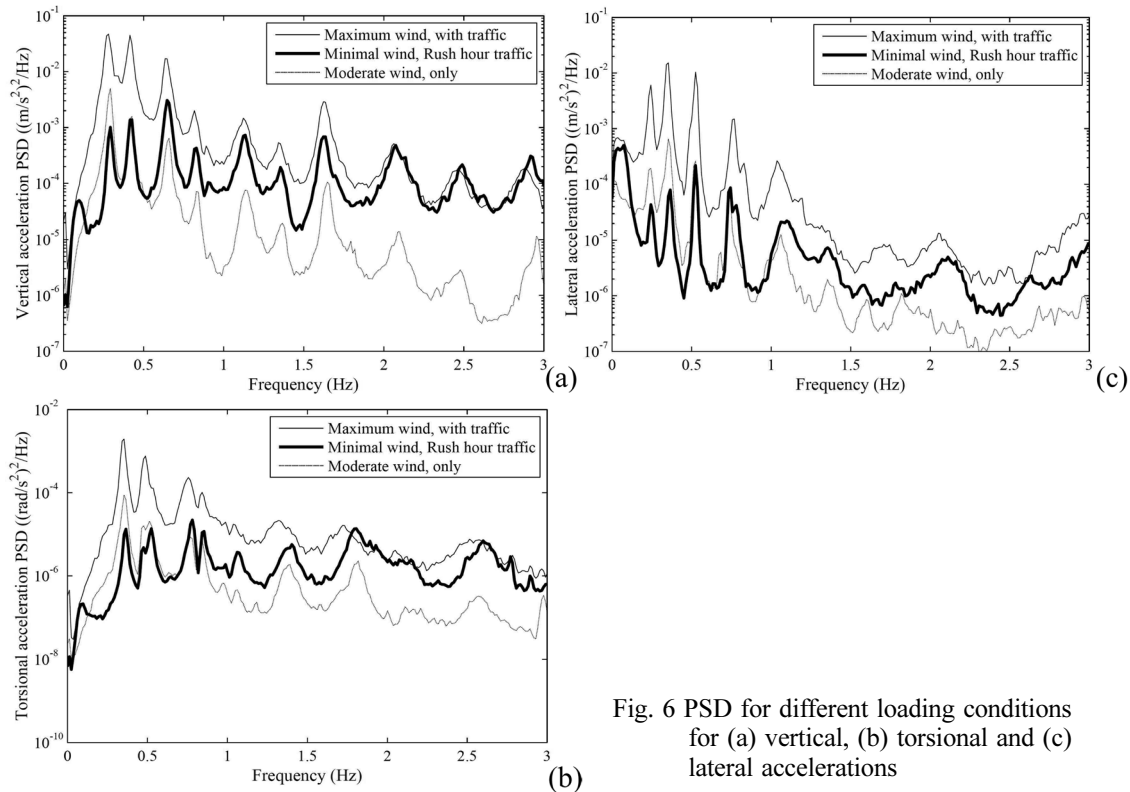


Fig. 6 PSD for different loading conditions for (a) vertical, (b) torsional and (c) lateral accelerations



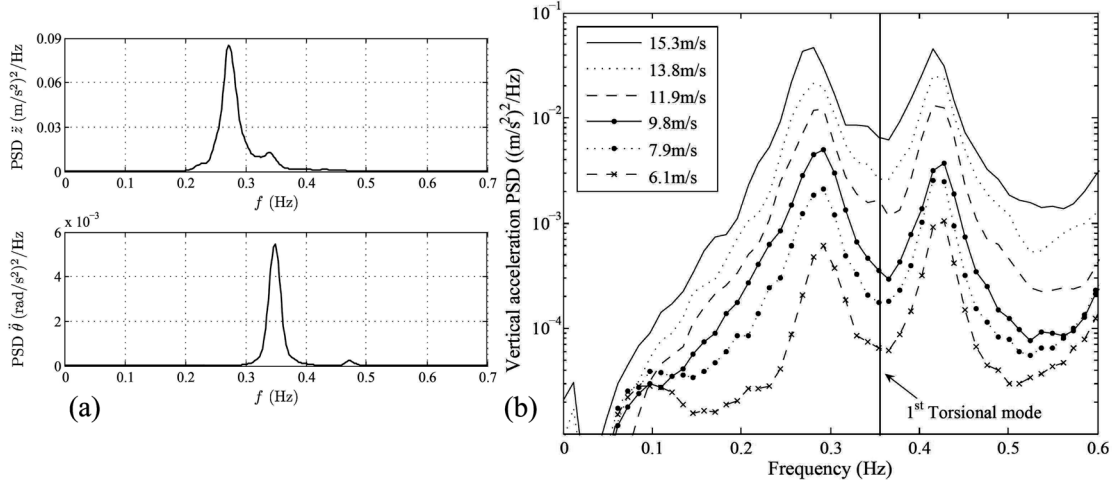


Fig. 7 (a) PSDs of filtered data for first vertical and torsional modes for the maximum wind speed record and (b) the evolution of the coupling action is evident in the vertical PSD for wind speeds above 11m/s

accelerations showed similar evidence, with the value around 0.35 Hz rising to approximately 0.6 in high winds compared with typical values below 0.4 in low winds and at other frequencies. It is also worth noting that the second pair of modes showed a similar tendency for coupling action in strong winds, due to their similar shapes (first symmetric mode of each type) and close natural frequencies (ratio  $0.424/0.498 \approx 0.85$ ). The next section discusses in more detail the identification of the CSB flutter derivatives, so as to be able to quantify the observed coupling signs and the potential for flutter.

## 5. Flutter derivatives

### 5.1 Flutter Analysis

According to the semi-empirical Selberg (1963) equation for bridge sections resembling flat plates, an estimate for the flutter speed is given by

$$\frac{U_F}{f_{\theta\theta} B} = C \sqrt{\frac{r_g m}{\rho B^3} \left[ 1 + \left( \frac{f_{z0}}{f_{\theta\theta}} \right)^2 \right]} \quad (2)$$

where  $U_F$  is the flutter speed,  $B$  the deck width,  $r_g$  the radius of gyration,  $m$  the total mass per unit length,  $C$  a constant depending on the mode shape similarity and the section's behavioural resemblance to flat plate,  $\tilde{n}$  the air density and  $f_{z0}$  and  $f_{\theta\theta}$  the still air vertical and torsional natural frequencies. For the first pair of natural frequencies, described above, and for  $\rho = 1.2 \text{ kg/m}^3$ ,  $B = 9.46 \text{ m}$ ,  $C = 2$  and the approximate values  $m = 5370 \text{ kg/m}$ ,  $r_g = 4 \text{ m}$ , the flutter onset speed is estimated as 18 m/s, or 14m/s using the more conservative variant of Eq. (2) in the British design rules (British Standards Institution 2009). These are very low values. This is due to the low torsional natural frequency, the

close neighbourhood of vertical and torsional modes, their shape affinity and the low mass per unit length. However, the bridge cross-section is not a flat plate and the uncertainty in the value of  $C$  means Eq. (2) is not very reliable, but it gives a first approximation for the flutter speed that is within the range of wind speeds potentially occurring at the site.

For evaluating the flutter behaviour the classical 2D formulation of Scanlan *et al.* (1997) is adopted, where aeroelastic forces are taken as a linear combination of the modal displacements and velocities appropriately multiplied with the so-called flutter derivatives. The motion-dependent lift and pitching moment  $L_{ae}$  and  $M_{ae}$  are hence given by

$$L_{ae} = \frac{1}{2}\rho U^2 B \left[ kH_1^* \frac{\dot{z}}{U} + kH_2^* \frac{B\dot{\theta}}{U} + k^2 H_3^* \theta + k^2 H_4^* \frac{z}{B} \right] \quad (3)$$

$$M_{ae} = \frac{1}{2}\rho U^2 B^2 \left[ kA_1^* \frac{\dot{z}}{U} + kA_2^* \frac{B\dot{\theta}}{U} + k^2 A_3^* \theta + k^2 A_4^* \frac{z}{B} \right] \quad (4)$$

where  $U$  is wind speed,  $z$  vertical displacement,  $\theta$  rotation and  $k = \omega B/U = 2\pi/U_r$  is reduced frequency, which is proportional to the number of oscillation cycles during the free stream flow passage over the width  $B$ .

In this 2D formulation, the motion-induced drag force,  $D_{ae}$ , and the effects of the along-wind motion are ignored. Under the assumption of low damping, as generally implied in the modelling of self-excited forces and conventional bridge flutter analysis, the flutter derivatives  $H_i^*$  and  $A_i^*$  with  $i=1,2,3,4$  become only functions of the reduced frequency  $k$  (or equivalently of the reduced wind speed  $U_r$ ), Chen (2007).

## 5.2 Identification method

A state space formulation of the dynamic problem was assembled using the Covariance Block Hankel Matrix Method (CBHM method), which is founded on the widely used Eigenvalue Realization Algorithm (ERA) described by Juang and Pappa (1985). The formulation in CBHM is identical to ERA with the exception that instead of the Markov parameters containing Impulse Response Functions (IRFs), covariance estimates of output measured random data are employed. Jakobsen (1995) first applied CHBM in the estimation of flutter derivatives from wind tunnel tests and the method has since found extensive application in aerodynamic applications and testing. Peeters and De Roeck (1999), Qin and Gu (2004) and Siringoringo and Fujino (2008) all describe the matrix derivation in detail. Brownjohn *et al.* (2010) found a relative advantage of the method over other operational modal analysis approaches. The method is based on the Singular Value Decomposition (SVD) and appropriate factorisation of a Hankel matrix built up by covariance estimates of the output time series (i.e., displacements or accelerations in this application). If  $y$  stands for the displacement matrix with  $z$  and  $\theta$  in rows in this case, then the unbiased sample cross-covariance matrix to be used in the Hankel matrix construction is given by

$$C_{yy}(i+n, i) = C_{yy}(n) = \frac{1}{N-n} \sum y(i+n)y^T(i), \quad n=0,1,\dots,l \quad (5)$$

where  $n$  is the number of sampling intervals for the discrete time delay  $n\Delta t$ ,  $N$  is the number of samples in the time series,  $l$  is the maximum number of lags considered and  $i$  is a counting index.

The biased estimate, which only differs in the use of the denominator  $N$  instead of  $N - n$ , can be used instead with negligible differences for long time records. The method typically assumes white noise loading but here it was attempted to also account for the actual shape of the loading spectra. Thus ordinary filtering was used on the response data in the frequency domain to account for the coloured lift and moment spectra given in Eq.(1). The main attribute that has to be taken care of is preserving after filtering the exact coupling between modes, that was artificially altered during filtering. The filter has monotony which is according to the loading spectra, so in any case it is expected that the side effect of distorted apparent coupling will be introduced. This is not realistic because buffeting action for each mode is uncorrelated to the coupling action due to self-excited forcing between modes. The remedy for this inconsistency was to modify the extracted coupled flutter derivatives with the ratio of the filter values so as to recover the initial coupling. Additional information on the performance of identifications with coloured noise and specific attributes on the relative sensitivity of each derivative are described by Jakobsen (1995). In practice, for the specific problem in hand the effect of the actual wind spectra on flutter derivatives was found to be insignificant.

The decomposition of the Hankel matrix recovers all parameters of the discrete time realisation. Knowing the modal stiffness and damping matrices for the in-wind and still air cases (pure structural stiffness and damping contributions) allow one to separate the flutter derivatives. The whole method (having the dimensionality for the problem already described as trivially two degrees-of-freedom) relies on the choice of two parameters; the length of the individual time record  $N$  and the number of time delays  $l$  for which the covariance matrix is evaluated and stored in the block Hankel matrix. The choice of both is investigated through a sensitivity analysis together with inspection of the time evolution of the auto and cross-covariance functions.

### 5.3 Application to the Clifton Suspension Bridge

The proximity of the fundamental vertical and torsional modes, as presented in Fig. 7, seems to encourage some coupling action, which could potentially be the initial signs of classical flutter. The PSDs in Fig. 7 imply some non-negligible values of  $H_2^*$  or  $H_3^*$  flutter derivatives since a coupling contribution is evidently occurring in the vertical PSD at the torsional frequency. The flutter derivative identification was performed in one case with recorded acceleration data and in another with displacements evaluated by double integration of the accelerations in hand. Both cases produced identical results.

For selection of the two foresaid identification parameters a range for time records from 10 minutes to 1 hour and, for the covariance function length, lags in the range of 10 to 40 seconds were used, preserving analogies with similar previous treatments. Example covariance functions, for moderately strong wind, are plotted against time lag in Fig. 8. As previously demonstrated by Jakobsen *et al.* (2003), the suitable number of maximum time lags is strongly influenced by the response character at different wind speeds. Higher wind speeds usually allow only a shorter

meaningful portion of the covariance function for accurately reproducing the two-degree-of-freedom interaction, e.g., due to high aerodynamic damping of the pure vertical response in case of streamlined box-girders. For our case an optimum set of values was found to be the combination of 15 minute records ( $N = 11 \times 2^{10}$ ) with time delays up to 20 seconds ( $l = 250$ ) with only weak sensitivity to  $l$ , probably due to the relatively small range of wind speeds.

Results for the CSB flutter derivatives are given in Fig. 9. No wind tunnel tests have been

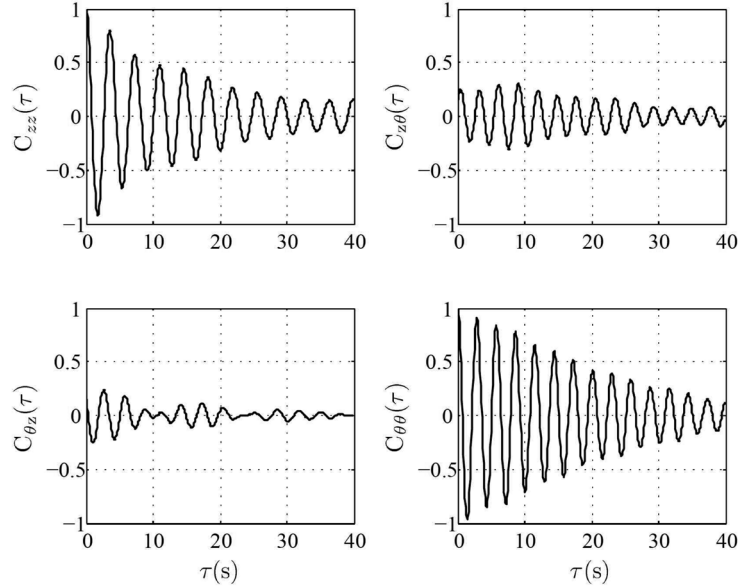


Fig. 8 Covariance functions for the combined two degrees-of-freedom plotted against time lag

undertaken on the CSB deck section, but where possible the site data are compared with available wind tunnel results of other deck cross-sections. Sign conventions for aerodynamic forces are as in Scanlan and Tomko (1971), i.e. with the lift force and the vertical displacement pointing downwards and the overturning moment and the rotation positive for the windward side of the bridge girder moving upwards. A sensitivity analysis on the measured wind characteristics, such as the turbulence and the angle of attack, proved not to be able to reproduce a clear picture of their effect. The identified trends of flutter derivatives remained unaltered, but data were insufficient to quantify a systematic impact of the investigated parameters.

Some of the derivatives in Fig. 9 appear to have an offset for still-air wind conditions. This has also been encountered in previous treatises (Jakobsen 1995), and here can mostly be attributed to effects such as the distortion from traffic effects, including both loading and variation in the mass distribution, as well as uncertainties in the modal masses and inaccuracies in the still-air structural matrices. For off-diagonal still-air values there is no remedy for this offset, which can be attributed to record noise and imperfections as explained by Hoen (2006). For diagonal values there was an attempt to minimize the offset for damping derivatives, since their absolute values are of greater importance.

Although the identified flutter derivatives are noisy, unsurprisingly for full-scale ambient data, some trends are apparent. Consistent with the observed bridge behaviour, the results indicate that, within the range of wind speeds recorded (maximum 15.3 m/s), the bridge is not susceptible to pure vertical or torsional flutter (so called “damping-driven flutter” as presented by Matsumoto *et al.* 1996), which was the reason for the famous Tacoma Narrows Bridge collapse. This is due to having close to negative  $H_1^*$  and  $A_2^*$  (direct damping derivatives), which will probably need a further increase in reduced wind speed to initiate such alarming behaviour, if indeed it does occur.

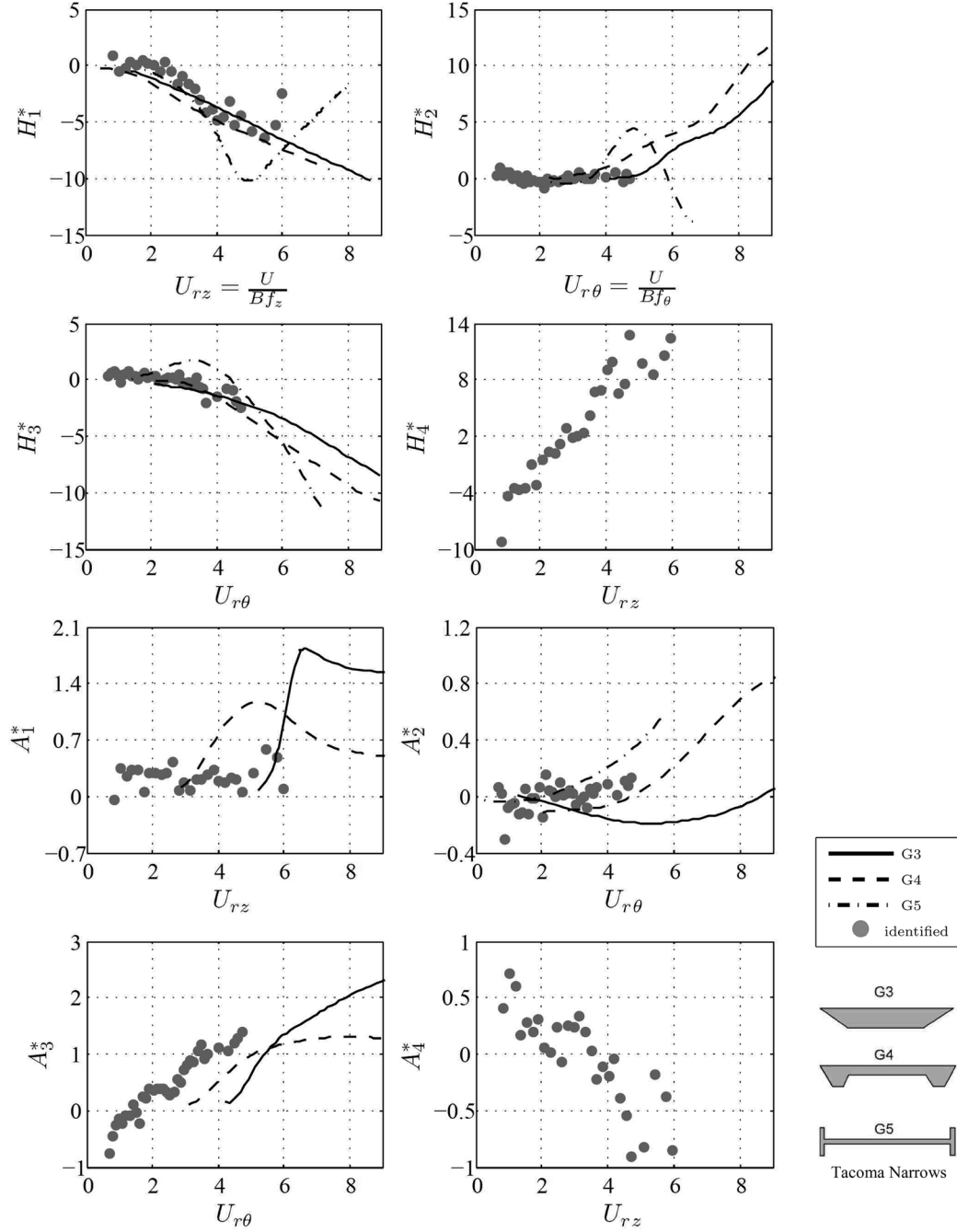


Fig. 9 Flutter derivatives of Clifton Suspension Bridge from full scale data, compared with wind tunnel extracted flutter derivatives for various cross-sections (after Scanlan and Tomko 1971 and adapted to Eqs. (3) and (4)).  $A_1^*$  and  $A_3^*$  for section G5 are negligible.  $H_4^*$  and  $A_4^*$  were not measured in the wind tunnel tests. Identified values correspond to binned and averaged identified values

However,  $H_1^*$  apparently shows a steep positive gradient near the highest wind speed recorded, suggesting it could become positive for higher wind speeds, possibly leading to flutter. This trend persists regardless of the selected identification parameters ( $N$  and  $l$ ), indicating it is not due to numerical errors, although the last few points in the figure are from averages over few records, so their accuracy may be limited. Fig. 10 shows the  $H_1^*$  flutter derivative estimates from each 15-minute record, before the averaging used for Fig. 9. Although only the last few points show the apparent positive gradient, these points depart significantly from the trend at lower reduced velocities and the differences are greater than the general scatter of points, implying there is a real effect. If this is confirmed, the effect of possible positive  $H_1^*$  (i.e., negative aerodynamic damping of vertical motion) could provide a feasible explanation for the occasional observations of large vibrations of the bridge in strong winds.

Dimensionally assessing the possibility of unstable motion in the pure vertical response, a value of  $H_1^* \approx 6$  needs to be reached. This estimate is based on the low amplitude structural damping ratio of  $\zeta = 3.3\%$  for the vertical mode (Macdonald 2008) and on the assumption that no beneficial amplitude-dependent increase in structural damping takes place. The structural damping is believed to be so high compared with modern suspension bridges because of the many joints in the structure, particularly the wrought iron suspension chains. Particular care had been taken in the system identification method to avoid bias errors that can often lead to erroneous overestimation of damping (Macdonald and Daniell 2005).

The  $H_2^*$  and  $H_3^*$  derivatives, which control the coupling from torsional to vertical motion, have small values. However at the higher wind speeds there is a noticeable growing negative trend in  $H_3^*$ , in line with the curves for other bridge profiles, which potentially explains the previously illustrated coupled spectra in Fig. 7. The relative influence of  $H_3^*$  for the evidently coupled response record translates to a vertical force approximately 1/10 of the peak restoring elastic force for the mode.

The evolution of  $H_4^*$  (aeroelastic direct vertical stiffness) reflects a reduction of vertical natural

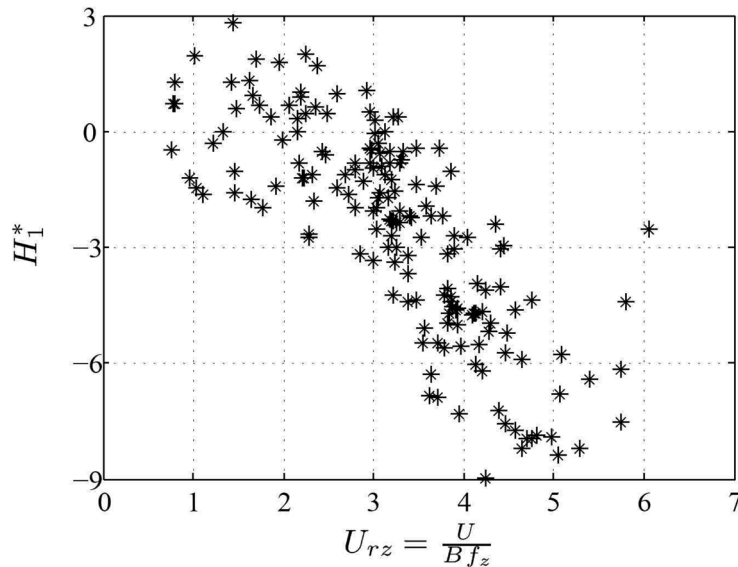


Fig. 10  $H_1^*$  flutter derivative from each 15-minute record

frequency with increasing wind speed, although this could alternatively be due to an amplitude dependence rather than the wind. Similarly  $A_3^*$  (aeroelastic direct torsional stiffness) illustrates a reduction in the torsional natural frequency with increasing wind speed. The highest identified values of  $A_3^*$  and  $H_4^*$  translate to an actual frequency drop of less than 7% in each case (c.f. variation of up to 4% from traffic (Macdonald 2008)). This reinforces previous observations that unlike airfoil flutter, for bridges with bluff sections aeroelasticity influences more the damping than the frequencies of the modes (see Scanlan and Tomko 1971 and Billah and Scanlan 1990), although in this study coupling was also apparent.

In any case, from the sections in hand a qualitative similarity was found with the bluff section of Tacoma Narrows. For a proper estimation of the critical flutter wind speed, through Complex Eigen-Value analysis (CEV), data inclusive of higher wind speeds are needed to extend the plots of Fig. 9. However, the Tacoma Narrows bridge failed under pure torsional flutter, due to  $A_2^*$  (negative torsional damping), so this flutter derivative is of particular interest. An estimate of its value at higher wind speeds can be obtained by utilising the relevant relationship between flutter derivatives from those proposed by Matsumoto (Matsumoto *et al.* 1996, Scanlan *et al.* 1997)

$$H_1^* = kH_3^*, H_4^* = -kH_2^*, A_1^* = kA_3^*, A_4^* = -kA_2^* \quad (6)$$

These suggested relationships are based on the assertion that twisting  $\theta$  and the apparent angle of attack associated with the bridge girder vertical velocity generate similar motion dependent forces. The relationships were found to yield an accurate match for many bluff cross sections. Thus Eq. (6) can be employed to investigate the possible extension of estimates of  $A_2^*$  to higher reduced wind speeds through the measured values of  $A_4^*$ . This can be achieved since the scaling of the reduced wind speed for  $A_4^*$  uses the lower vertical frequency giving higher reduced wind speeds than  $A_2^*$ , which is expressed in relation to the torsional frequency. Consequently it is possible to review the possibility of single-degree-of-freedom torsional flutter on the CSB. Fig. 11 shows the  $A_2^*$  data with the resulting additional points. Fitting the polynomial  $A_2^* = 0.12U_{r\theta}(0.3U_{r\theta} - 1)$  provides an estimate for pure torsional flutter at  $U_{r\theta} \approx 6.3$  i.e., at  $U \approx 21$  m/s. This estimate is based on the low amplitude structural damping estimate for the torsional mode of  $\zeta = 2.6\%$  (Macdonald 2008) and does not allow for a potential beneficial increase of structural damping with amplitude. Because of this, the uncertainty in the estimation of such small aerodynamic forces as presented by  $A_4^*$  and the uncertainty in the relationships in Eq. (6), the actual flutter wind speed could be higher, although results of this magnitude are still significantly below today's standards.

## 6. Conclusions

The measured full-scale wind-induced response of the Clifton Suspension Bridge has been analysed and flutter derivatives extracted using a subspace stochastic identification method. Although no wind tunnel tests have been performed on the particular deck section, the identified flutter derivatives showed some clear trends and some similar features as found in wind tunnel tests on the Tacoma Narrows Bridge section. For the historic Clifton Suspension Bridge, which has been in service for more than 140 years to date, the results indicated no problem within the range of wind speeds recorded (up to 15.3 m/s) although there were potential concerning trends in the flutter derivatives for higher wind speeds. Operation at wind speeds so close to potential instabilities

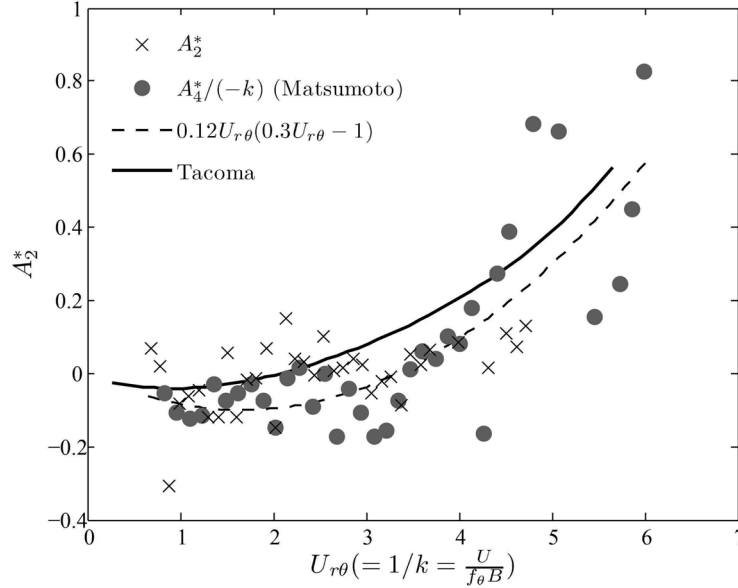


Fig. 11 Flutter derivative  $A_2^*$  with additional points from  $A_4^*$  as suggested by Matsumoto *et al.* (1996). The fitted polynomial, indicated by the broken line, was used for estimation of a pure torsional flutter wind speed

provided unique full-scale data of a bridge in this condition, making it an interesting case study and yielding clearer trends in the results than from any previous full-scale case study. Aeroelastic coupling was identified between the close first vertical and torsional modes. However, there appears to be a clearer potential hazard from single-degree-of-freedom instabilities, with the potential for negative aerodynamic damping caused by positive  $H_1^*$  or  $A_2^*$ . The behaviour of  $A_2^*$  at higher reduced velocities has been estimated based on the measured  $A_4^*$ , revealing an apparent trend. On this basis a wind speed estimation for pure torsional flutter has been made. Although uncertain, it is believed this is the first time this has been achieved based solely on the actual full-scale dynamic performance of a bridge.

This paper attempts to contribute to the literature on the analysis of aeroelastic effects from ambient vibration data on full-scale bridges, being one of very few similar studies. It shows the viability of extracting flutter derivatives from full-scale data, even in the presence of variable wind directions, turbulence intensity, traffic, etc., and also the potential value of system identification methods to validate full-scale performance of structures.

## Aknowledgements

The authors gratefully acknowledge the support of the Clifton Suspension Bridge Trust for the site tests and the support of EPSRC for the analysis under Macdonald's Advanced Research Fellowship. Flint & Neill, consultants for the bridge, are thanked for their constructive comments on the manuscript.



## References

- Barlow, W.H. (1867), "Description of the Clifton Suspension Bridge", *P. I. Civil Eng. Bridge Eng.*, **156**(1), 5-10.
- Bietry, J., Delaunay, D. and Conti, E. (1995), "Comparison of full-scale measurements and computation of wind effects on a cable-stayed bridge", *J. Wind Eng. Ind. Aerod.*, **57**(2-3), 225-235.
- Billah, K.Y. and Scanlan, R.H. (1990), "Resonance, Tacoma Narrows bridge failure, and undergraduate physics textbooks", *Am. J. Phys.*, **59**(2), 118-124.
- British Standards Institution (2009), "Published Document: Background information to the National Annex to BS EN 1991-1-4 and additional guidance", BS PD 6688-1-4:2009.
- Brownjohn, J.M.W. (1994), "Estimation of damping in suspension bridges", *P. I. Civil Eng. Str.B.*, **104**, 401-415.
- Brownjohn, J.M.W., Magalhaes, F., Caetano, E. and Cunha A. (2010), "Ambient vibration retesting and operational modal analysis of the Humber Bridge", *Eng. Struct.*, **32**(8), 2003-2018.
- Chen, X. (2007), "Improved understanding of bimodal coupled bridge flutter based on closedform solutions", *J. Struct. Eng-ASCE*, **133**(1), 22-31.
- Costa, C. and Borri, C. (2007), "Full-scale identification of aeroelastic parameters of bridges", *Proceedings of the 12th Int. Conf. Wind Eng.*, Cairns, Australia.
- Farquharson, F.B., Smith, F.C. and Vincent, G.S. (1950-1954), Aerodynamic stability of suspension bridges with special reference to the Tacoma Narrows Bridge (five parts), Engineering Experiment Station, University of Washington, USA.
- Ge, Y.J. and Tanaka, H. (2002), "Aerodynamics of long-span bridges under erection", *J. Struct. Eng.*, **126**, 1404-1412.
- Hoen, C. (2006) "Subspace identification of modal coordinate time series", *Proceedings of the 24th Int. Modal Anal.Conf.*, IMAC XXIV, St.Louis, USA.
- Jakobsen, J.B. and Hjorth-Hansen, E. (1995), "Determination of the aerodynamic derivatives by a system identification method", *J. Wind Eng. Ind. Aerod.*, **57**(2-3), 295-305.
- Jakobsen, J.B. and Larose, G.L. (1999), "Estimation of aerodynamic derivatives from ambient vibration data", *Proceedings of the 10th Int. Conf. Wind Eng.*
- Jakobsen, J.B., Savage, M.G. and Larose, G.L. (2003), "Aerodynamic derivatives from the buffeting response of a flat plate model with stabilizing winglets", *Proceedings of the 11th Int. Conf. Wind Eng.*, Lubbock, Texas.
- Jensen, J.L., Larsen, A., Andersen, J.E. and Vejrum, T. (1999), "Estimation of structural damping of Great Belt suspension bridge", *Proceedings of the 4th Euro. Conf. Struct. Dyn.* (Eurodyn '99), Prague.
- Jones, N.P., Scanlan, R.H., Sarkar, P.P. and Singh, L. (1995), "The effect of section model details on aeroelastic parameters", *J. Wind Eng. Ind. Aerod.*, **54-55**, 45-53.
- Juang, J.N. and Pappa, R.S (1985), "An eigensystem realization algorithm for modal parameter identification and model reduction", *J. Guid. Control Dynam.*, **8**(5), 620-627.
- Kumarasena, T., Scanlan, R.H. and Morris G.R. (1989), "Deer Isle: Efficacy of stiffening system", *J. Struct. Eng-ASCE*, **115**(9), 2297-2312.
- Kumarasena, T., Scanlan, R.H. and Morris G.R. (1989), "Deer Isle: Field and computed vibrations", *J. Struct. Eng-ASCE*, **115**(9), 2313-2328.
- Littler, J.D. (1992), "Ambient vibration tests on long span suspension bridges", *J. Wind Eng. Ind. Aerod.*, **42**(1-3), 1359-1370.
- Macdonald, J.H.G. (2003), "Evaluation of buffeting predictions of a cable-stayed bridge from fullscale measurements", *J. Wind Eng. Ind. Aerod.*, **91**(12-15), 1465-1483.
- Macdonald, J.H.G. and Daniell, W.E. (2005), "Variation of modal parameters of a cable-stayed bridge identified from ambient vibration measurements and FE modelling", *Eng. Struct.*, **27**(13), 1916-1930.
- Macdonald, J.H.G. (2008), "Pedestrian-induced vibrations of the Clifton Suspension Bridge, UK", *P. I. Civil Eng. Bridge Eng.*, **161**(2), 69-77.
- Matsumoto, M., Koboyashi, Y. and Shirato, H. (1996), "The influence of aerodynamic derivatives on flutter", *J. Wind Eng. Ind. Aerod.*, **60**, 227-239.
- Matsumoto, M., Nakajima, N., Taniwaki, Y. and Shijo, R. (2001), "Grating effect on flutter instability", *J. Wind Eng. Ind. Aerod.*, **89**(14-15), 1487-1498.
- Nagayama, T., Abe, M., Fujino, Y. and Ikeda, K. (2005), "Structural identification of a nonproportionally

- damped system and its application to a full-scale suspension bridge”, *J. Struct. Eng-ASCE*, **131**(10), 1536-1545.
- Peeters, B. and De Roeck, G. (1999), “Reference-based stochastic subspace identification for output-only modal analysis”, *Mech. Sys. Signal Pr.*, **13** (6), 855-878.
- Qin, X.R and Gu, M. (2004), “Determination of flutter derivatives by stochastic subspace identification technique”, *Wind Struct.*, 7(3), 173-186.
- Scanlan, R.H. and Tomko, J.J. (1971), “Airfoil and bridge flutter derivatives”, *J. Eng. Mech. Div-ASCE*, **97**(6), 1717-1737.
- Scanlan, R.H., Jones, N.P. and Singh, L. (1997), “Inter-relations among flutter derivatives”, *J.Wind Eng. Ind. Aerod.*, **69-71**, 829-837.
- Selberg, A. (1963), “Aerodynamic effects on suspension bridges”, *Int. Conf. Wind Eff. Build. Struct.*, Vol. **2**, 462-479.
- Siringoringo, D.M and Fujino, Y. (2008), “System identification of suspension bridge from ambient vibration response”, *Eng. Struct.*, **30**(2), 462-477.