Wave propagation of functionally graded anisotropic nanoplates resting on Winkler-Pasternak foundation

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Abstract. This work deals with the size-dependent wave propagation analysis of functionally graded (FG) anisotropic nanoplates based on a nonlocal strain gradient refined plate model. The present model incorporates two scale coefficients to examine wave dispersion relations more accurately. Material properties of FG anisotropic nanoplates are exponentially varying in the z-direction. In order to solve the governing equations for bulk waves, an analytical method is performed and wave frequencies and phase velocities are obtained as a function of wave number. The influences of several important parameters such as material graduation exponent, geometry, Winkler-Pasternak foundation parameters and wave number on the wave propagation of FG anisotropic nanoplates resting on the elastic foundation are investigated and discussed in detail. It is concluded that these parameters play significant roles on the wave propagation behavior of the nanoplates. From the best knowledge of authors, it is the first time that FG nanoplate made of anisotropic materials is investigated, so, presented numerical results can serve as benchmarks for future analysis of such structures.

Keywords: wave propagation; functionally graded anisotropic materials; nonlocal strain gradient theory; four variable shear deformation refined plate theory; elastic foundation

1. Introduction

Classical plate theory CPT as the simplest plate theory, cannot model thick/moderately thin plates accurately because of several assumptions for simplifications used in this theory (Lagnese et al. 2015, Farokhi and Ghayesh 2015). On the other hand, first order plate theory (FSDT) with less simplifications with two main unknowns give us better results for thick/moderately thin plates although the results still have some inaccuracy. Besides, this theory also needs shear correction factor. For solving these two problems, several higher order shear deformation theories (HSDTs) have been developed. The problem with these HSDTs is the complexity of them. Some of them have more than six main unknowns which cause several long equations. In this research, we choose one of the theories which has some of the advantages and disadvantages mentioned above. This theory, known as refined plate theory (RPT), does not need any shear correction factor and it is simple (i.e. with only two unknowns) with results more accurate than CPT which can be a candidate for engineering applications, (see some applications of HSDTs and RPTs in Refs. (Shimpi and Patel 2006, Mehar et al. 2018b, Katariya et al. 2017a, Dutta et al. 2017, Sahoo et al. 2017, Singh and Panda 2017, Mehar et al. 2017, Mahapatra et al. 2017, Mehar and Panda 2017a, Mehar et al. 2018c, Hirwani and Panda 2019, Hirwani and Panda 2018, Dash et al. 2018b, Mehar et al. 2018a, Hirwani et al. 2018, Mehar and Panda 2018, Katariya et al. 2018, Katariya et al. 2018b, Mehar and Panda 2018, Katariya et al. 2018b, Mehar and Panda 2018b, Mehar and Panda 2018b, Sayyad and Ghugal 2018b.

In the past decades, with the development of nanotechnology, we need to investigate the nanostructures that old models such as CPT, HSDTs and RPTs, etc. cannot analyzed the mechanical behavior of such structures accurately. Hence, to consider the size-dependent effect, nanostructures usually modelled based on the molecular dynamics (MD) simulation and non-classical theories. Eringen nonlocal model (Eringen 1983) is one of the nonclassical theories in which stress at a reference point are assumed to depend not only on the strains at the reference point but also on the strains at all other points in the body. As an example of the merits of this theory, one can mention the non-monotonicity of dispersion between phase velocity and wave number predicted by MD simulation. Several studies have been done based on this size-dependent model (Karami et al. 2018f, Karami et al. 2019b, Shahsavari et al. 2018a, Shahsavari et al. 2017, Shahsavari and Janghorban 2017, Apuzzo et al. 2017, Barretta et al. 2018, Romano and Barretta 2017, Romano et al. 2017, Shahsavari et al. 2018e, Karami et al. 2018l). Free vibration analysis of FG nanoplates resting on elastic foundation using a nonlocal zeroth-order shear deformation theory were studied by

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(Bounouara et al. 2016). (Ebrahimi and Barati 2016) investigated the free vibration analysis of FG nanobeams third-order shear deformation plate theory and nonlocal elasticity theory. Bending analysis of bi-directional FG Euler-Bernoulli nano-beams using Eringen's non-local elasticity theory were presented by (Nejad and Hadi 2016). Based on the first-order plate theory in conjunction with nonlocal elasticity theory, guided wave propagation analysis of fully-clamped porous FG nanoplates were studied by (Karami et al. 2018a) for the first time. As can be seen, there are numerous applications on the mechanical analysis of nanostructures based on Eringen nonlocal model (Moradweysi et al. 2018, Ehyaei et al. 2017, Khetir et al. 2017, Chaht et al. 2015, Simsek 2011, Rahmani et al. 2017, Mirjavadi et al. 2017, Bellifa et al. 2017, Sobhy 2017). Nevertheless, nonlocal elasticity theory cannot always predict the manner of materials at nanoscale well, and therefore the sufficiency of nonlocal elasticity theory for predicting physical behavior is questionable for some special cases. Among the examples that can be mentioned there are may be some limitations to its capability of identifying size-dependent stiffness (Ma et al. 2008, Lim et al. 2015). Moreover, still, the problem remains why, in some cases, the results of bending of nonlocal nanobeams are ineffective at all (Challamel and Wang 2008). Besides, it can be seen that the stiffness enhancement effects seen from experimental data and the strain gradient model (Lam et al. 2003) cannot be included well by adopting the Eringen's nonlocal elasticity theory. It is worth noting that in other size-dependent studies, we observe the importance of non-classical boundary conditions, but in this theory we see that in some cases the same classical boundary conditions suffice, which is a question for some researchers (Volokh and Hutchinson 2002, Polizzotto 2003). Another one is the gradient model of elasticity (Mindlin 1964, Aifantis 1992) which considers additional higher-order strain gradient terms by adopting the fact that the materials should be modelled as atoms with higher-order deformation mechanism at nanoscale rather than set of points. A modified gradient elasticity theory, which states the strain energy density should be considered as a function of both the strain tensor conjugated with stress tensor and the curvature tensor conjugated with couple stress tensor were presented by (Yang et al. 2002). Initial stress affected wave propagation behavior of nanoplates were investigated by (Nami and Janghorban 2014) using gradient elasticity theory. (Karami and Janghorban 2016) investigated the wave propagation in rectangular nanoplates via gradient elasticity theory and two-variable RPT.

With regard to the above literature, it is easy to conclude that the nonlocal and gradient models of elasticity explain two entirely different physical characteristics of materials at small scale. More recently, to study the influence of the two small scale parameters on the structural responses, (Lim *et al.* 2015) presented the nonlocal strain gradient theory with the approach of the thermodynamics framework to consider both of the nonlocal and gradient parameters into a single theory. Many researches have been carried out on the mentioned theory (She *et al.* 2018c, She *et al.* 2018a, Karami *et al.* 2018d, Barati 2017, Zhu and Li 2017, Xiao *et*

al. 2017, She et al. 2019, She et al. 2017, Shafiei and She 2018, She et al. 2018b, Karami et al. 2018e, Shahsavari et al. 2018d, Karami et al. 2018j, Karami et al. 2018g, Karami et al. 2018k, Karami et al. 2018b, Karami et al. 2018i, Shahsavari et al. 2018b, Shahsavari et al. 2018c, Karami et al. 2019a, Karami et al. 2018c, Ansari et al. 2011, Sahmani et al. 2018b, Sahmani et al. 2018a, Karami and Karami 2019, Nami and Janghorban 2015, Karami and Janghorban 2019). Li et al. (2016), reported that a good matching of the dispersive curve of molecular dynamic simulations can be obtained with nonlocal beams and illustrated that the stiffness softening effects or the stiffness enhancement effects can be obtained depend on the values of the nonlocal parameter and the strain gradient length scale parameter. Also, in another study a good agreement is achieved for wave dispersion curves of graphene and experimental data using nonlocal strain gradient second-order plate theory by (Karami et al. 2018h). Based on the nonlocal strain gradient theory, (Li and Hu 2015) investigated the stability analysis of nonlinear nanobeams and showed that the stiffness softening effects or the stiffness enhancement effects can be obtained depend on the values of the nonlocal parameter and the strain gradient length scale parameter. The nonlocal strain gradient theory can be considered among other theories as one of the most accurate modeling for the study of nanostructures with considering both nonlocal and gradient effects.

FGMs have received a considerable attention in several engineering applications due to its wonderful properties in different environments. This type of materials is characterized by its properties that vary continuously or exponentially through a direction like thickness. Generally, FG structures are made of isotropic or anisotropic materials in which to manufacturing FG anisotropic materials, transport phenomena are used to create compositional and microstructural gradients during the production of a component.

A few number of studies have been conducted on the static and dynamic behavior of shell and plates in micro/nano-scale made of different anisotropic materials (i.e., anisotropic and FG anisotropic). Vibrational behavior of three-dimensional anisotropic layered composite nanoplates using modified couple-stress theory were studied by (Guo et al. 2017). (Sahmani and Fattahi 2017) investigated the nonlocal anisotropic shear deformable plate model for uniaxial instability of 3D metallic carbon nanosheets. In their study a calibration is presented for nonlocal anisotropic shear deformable plate using molecular dynamic simulation. The wave propagation analysis of anisotropic plate using trigonometric shear deformation theory were studied by (Aminipour and Janghorban 2017). (Karami et al. 2017) presented three dimensional nonlocal strain gradient plate model for wave propagation behavior of anisotropic nanoplates under the influences of triaxial magnetic field. (Aminipour et al. 2018) investigated the wave propagation in FG anisotropic shells based on a new model for wave analysis.

In the current study, due to the lack of any study on the mechanics of graded nanoplates made of anisotropic materials, we prepared an article to investigate the wave

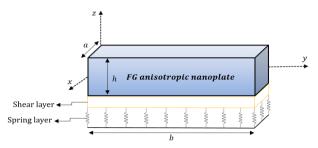


Fig. 1 Geometry of FG anisotropic nanoplate rested on an elastic foundation

propagation analysis of FG anisotropic nanoplate. Fourvariable RPT without any shear correction factor is used to model the plate as a continuum model. For more accurate design, the proposed modeling of nanoplates incorporates a nonlocal stress field parameter as well as a length scale parameter related to strain gradient. Thus, stiffness enhancement or reduction observed in nanostructures are considered. Governing equations are obtained using Hamilton's principle. By solving the governing equations using an analytical method, wave frequencies as well as phase velocities of the FG anisotropic nanoplate are obtained as a function of wave number. The results show that wave characteristics of FG anisotropic nanoplate is significantly influenced by the nonlocality, strain gradient parameter, material composition, Winkler-Pasternak foundation and geometrical parameters. Obtained results can be used as benchmark results in the analysis of FG anisotropic nanoplates modeled by nonlocal and straingradient theories.

2. Theoretical formulation

2.1 The material properties of FG anisotropic elastic nanoplates

Rectangular nanosize plate made of FG anisotropic materials is considered here (Fig. 1). A Cartesian coordinate system (x, y, z) is assumed; a and b are length and width of plate and h is the plate thickness.

The constitutive relation for each material can be written as

$$\sigma = C \varepsilon \tag{1}$$

where σ and ε respectively, denote stress and strain terms and C denotes the elastic constant that is different in various structures. For FGMs with exponentially varying material properties in the z-direction, the elastic stiffness matrix in Eq. (1) can be expressed as (Pan 2003, Karami *et al.* 2018c)

$$C(z) = C^{0} \exp(\lambda z) \tag{2}$$

where λ is the exponential factor defined the material gradient's degree in the thickness direction. It is also mentioned that the exponential factor λ has the dimension 1/L, in which L defined as the characteristic length of the problem. In the current study, to investigate the wave

propagation of graded nano-size plates, a hexagonal system (beryllium crystal) is used. It has an axis of symmetry such that a rotation of the crystal through 60° about that axis brings the space lattice into coincidence with its original configuration (Batra *et al.* 2004). The elastic constants are,

$$C^{0} = \begin{bmatrix} 298.2 & 27.7 & 0 & 0 & 0\\ 298.2 & 0 & 0 & 0\\ & 165.5 & 0 & 0\\ & \text{sym.} & 165.5 & 0\\ & & 135.3 \end{bmatrix} \times 10^{9} \text{Pa}$$
 (3)

and the density of beryllium equals to $\rho = 1850 \frac{\text{kg}}{\text{m}^3}$.

2.2 Nonlocal strain gradient elasticity

The total nonlocal strain gradient stress tensor can be expressed as below (Lim *et al.* 2015),

$$\sigma_{ij} = \sigma_{ij}^{(0)} - \nabla \partial \sigma_{ij}^{(1)} \tag{4}$$

in which $\sigma_{ij}^{(0)}$, $\sigma_{ij}^{(1)}$ in order are related to strain \mathcal{E}_{ij} and strain gradient $\nabla \mathcal{E}_{ii}$ and are defined as

$$\sigma_{ij}^{(0)} = \int_{0}^{L} C_{ijkl} \alpha_{0}(x, x', e_{0}a) \varepsilon_{kl}'(x') dx'$$
 (5)

$$\sigma_{ij}^{(1)} = l^2 \int_0^L C_{ijkl} \alpha_1(x, x', e_1 a) \nabla \varepsilon_{kl}'(x') dx'$$
 (6)

where C_{ijkl} are the elastic constants and e_0a and e_1a consider the influence of nonlocal stress field and l is the strain gradient small scale parameter which defines the effects of higher order strain gradient stress field. As the conditions by Eringen are satisfied by the nonlocal functions $\alpha_0(x,x',e_0a)$ and $\alpha_0(x,x',e_1a)$, the constitutive equation of nonlocal strain gradient theory can be written in following format

$$\begin{bmatrix}
1 - (e_1 a)^2 \nabla^2 \end{bmatrix} \begin{bmatrix}
1 - (e_0 a)^2 \nabla^2 \end{bmatrix} \sigma_{ij}
= C_{ijkl} \begin{bmatrix}
1 - (e_1 a)^2 \nabla^2 \end{bmatrix} \varepsilon_{kl} - C_{ijkl} l^2 \begin{bmatrix}
1 - (e_0 a)^2 \nabla^2 \end{bmatrix} \nabla^2 \varepsilon_{kl}$$
(7)

in which ∇^2 is represents Laplacian operator in Cartesian coordinates. Supposing $e_1 = e_0 = e$ and discarding terms of order $O(\nabla^2)$, the general constitutive relation in Eq. (7) can be rewritten as (Lim *et al.* 2015)

$$\left[1 - (ea)^2 \nabla^2\right] \sigma_{ij} = C_{ijkl} \left[1 - l^2 \nabla^2\right] \varepsilon_{kl}$$
 (8)

The equivalent form of Eq. (8) is presented as

$$L_{\mu}\sigma_{ij} = C_{ijkl}L_{l}\varepsilon_{kl} \tag{9}$$

in which the linear operators are defined as

$$L_{\mu} = (1 - \mu^2 \nabla^2), L_{t} = (1 - l^2 \nabla^2)$$
 (10)

where $\mu = ea$.

2.3 Refined plate theory (RPT)

According to the plate theories, the refined model initially proposed by (Shimpi 2002) is widely used as a reliable theory in which no shear correction factor is required. In Shimpi's theory, the displacement field, for $z \in [-h/2;h/2]$, is defined as

$$u(x,y,z,t) = u_0(x,y,t) - z \frac{\partial w_b}{\partial x} - f(z) \frac{\partial w_s}{\partial x}$$
 (11)

$$v(x,y,z,t) = v_0(x,y,t) - z \frac{\partial w_b}{\partial y} - f(z) \frac{\partial w_s}{\partial y}$$
 (12)

$$w(x, y, z, t) = w_b(x, y, t) + w_s(x, y, t)$$
(13)

where u_0 , v_0 , w_b and w_s reperesent, respectively, in-plane displacements and the bending and shear transverse displacements. The shape function of transverse shear deformation is considered as

$$f(z) = -\frac{z}{4} + \frac{5z^3}{3h^2} \tag{14}$$

Nonzero strains according to the four-variable RPT can be written as

$$\begin{cases}
\mathcal{E}_{x} \\
\mathcal{E}_{y} \\
\gamma_{xy}
\end{cases} = \begin{cases}
\mathcal{E}_{x}^{0} \\
\mathcal{E}_{y}^{0} \\
\gamma_{xy}^{0}
\end{cases} + z \begin{cases}
K_{x}^{b} \\
K_{y}^{b} \\
K_{xy}^{b}
\end{cases} + f \begin{cases}
K_{x}^{s} \\
K_{y}^{s} \\
K_{xy}^{s}
\end{cases}$$

$$\begin{cases}
\gamma_{yz} \\
\gamma_{xz}
\end{cases} = g \begin{cases}
\gamma_{yz}^{s} \\
\gamma_{xz}^{s}
\end{cases}, g = 1 - \frac{\partial f}{\partial z}$$
(15)

where

$$\begin{cases}
\mathcal{E}_{x}^{0} \\
\mathcal{E}_{y}^{0} \\
\gamma_{xy}^{0}
\end{cases} = \begin{cases}
\frac{\partial u_{0}}{\partial x} \\
\frac{\partial v_{0}}{\partial y} \\
\frac{\partial u_{0}}{\partial y} + \frac{\partial v_{0}}{\partial x}
\end{cases}, \begin{cases}
k_{x}^{b} \\
k_{y}^{b} \\
k_{xy}^{b}
\end{cases} = \begin{cases}
-\frac{\partial^{2} w_{b}}{\partial x^{2}} \\
-\frac{\partial^{2} w_{b}}{\partial y^{2}} \\
-2\frac{\partial^{2} w_{b}}{\partial x \partial y}
\end{cases}$$

$$\begin{cases}
k_{x}^{s} \\
k_{xy}^{s}
\end{cases} = \begin{cases}
-\frac{\partial^{2} w_{s}}{\partial x^{2}} \\
-\frac{\partial^{2} w_{s}}{\partial y^{2}} \\
-2\frac{\partial^{2} w_{s}}{\partial y \partial y}
\end{cases}, \begin{cases}
\gamma_{xz}^{s} \\ \gamma_{yz}^{s}
\end{cases} = \begin{cases}
\frac{\partial w_{s}}{\partial x} \\
\frac{\partial w_{s}}{\partial y}
\end{cases}$$
(16)

Also, the extended Hamilton principle introduces as

$$\int_0^t \delta(U - T + V) dt = 0 \tag{17}$$

where U is strain energy, T is kinetic energy and V is work of external (applied) forces. With substituting the corresponding terms, the first variation of strain energy can be written as

$$\delta U = \int_{V} \left[\sigma_{x} \delta \varepsilon_{x} + \sigma_{y} \delta \varepsilon_{y} + \tau_{yz} \varepsilon_{yz} + \tau_{xz} \varepsilon_{xz} + \tau_{xy} \varepsilon_{xy} \right] dV$$

$$= \int_{0}^{L} \left[N_{x} \delta \varepsilon_{x}^{0} + N_{y} \delta \varepsilon_{y}^{0} + N_{xy} \delta \gamma_{xy}^{0} + M_{x}^{b} \delta k_{x}^{b} + M_{y}^{b} \delta k_{y}^{b} + M_{xy}^{b} \delta k_{xy}^{b} + M_{xy}^{s} \delta k_{xy}^{s} + M_{xy}^{s} \delta k_{xy}^{s} + Q_{yz}^{s} \gamma_{yz}^{0} + Q_{xz}^{s} \gamma_{yz}^{0} \right] dx = 0$$

$$(18)$$

where N, M, and Q are the stress resultants which are defined as

$$\begin{cases}
\{N\} \\
\{M^{b}\} \\
\{M^{s}\} \\
\{M^{s}\} \\
\} =
\begin{bmatrix}
[A] & [B] & [B^{s}] \\
[B] & [D] & [D^{s}] \\
[B^{s}] & [D^{s}] & [H^{s}] \\
\end{bmatrix}
\begin{cases}
\mathcal{E}^{0} \\
k^{b} \\
k^{s}
\end{cases}$$

$$\begin{cases}
\{Q_{xz}\} \\
\{Q_{yz}\} \\
\{Q_{yz}\} \\
\} =
\begin{bmatrix}
A_{44}^{s} & 0 \\
0 & A_{55}^{s}
\end{bmatrix}
\begin{cases}
\gamma_{xz}^{s} \\
\gamma_{yz}^{s}
\end{cases}$$
(19)

in which

$$(A,B,B^{s},D,D^{s},H^{s}) = \int_{-\frac{h}{2}}^{\frac{h}{2}} C_{ij}(1,z,f,z^{2},zf,f^{2})dz$$
 (20)

and

$$(A_{44}^{s}, A_{55}^{s}) = \int_{-\frac{h}{2}}^{\frac{h}{2}} (C_{44}, C_{55}) g^{2} dz$$
 (21)

The first variation of work done by applied forces can be stated as

$$\delta V = \int_{-h/2}^{h/2} \int_{A} \left[-k_{W} \delta(w_{b} + w_{s}) + k_{P} \left(\frac{\partial (w_{b} + w_{s})}{\partial x} \frac{\partial \delta(w_{b} + w_{s})}{\partial x} \right) + \frac{\partial (w_{b} + w_{s})}{\partial y} \frac{\partial \delta(w_{b} + w_{s})}{\partial y} \right] dA dz$$
(22)

where k_W and k_P represent, respectively, the linear and shear coefficient of elastic foundation. The variation of kinetic energy according to the present theory can be achieved as

$$\begin{split} &\delta K = \int_{-h/2}^{h/2} \int_{A} \left[\dot{u}_{0} \delta \dot{u}_{0} + \dot{v}_{0} \delta \dot{v}_{0} + \dot{w} \delta \dot{w} \right] dA dz \\ &= \int_{A} \left\{ I_{0} \left(\dot{u}_{0} \delta \dot{u}_{0} + \dot{v}_{0} \delta \dot{v}_{0} + \left(\dot{w}_{b} + \dot{w}_{s} \right) (\delta \dot{w}_{b} + \delta \dot{w}_{s} \right) \right) \\ &- I_{1} \left(\dot{u}_{0} \frac{\partial \delta \dot{w}_{b}}{\partial x} + \frac{\partial \dot{w}_{b}}{\partial x} \delta \dot{u}_{0} + \dot{v}_{0} \frac{\partial \delta \dot{w}_{s}}{\partial y} + \frac{\partial \dot{w}_{s}}{\partial y} \delta \dot{v}_{0} \right) \\ &- J_{1} \left(\dot{u}_{0} \frac{\partial \delta \dot{w}_{s}}{\partial x} + \frac{\partial \dot{w}_{s}}{\partial x} \delta \dot{u}_{0} + \dot{v}_{0} \frac{\partial \delta \dot{w}_{s}}{\partial y} + \frac{\partial \dot{w}_{s}}{\partial y} \delta \dot{v}_{0} \right) \\ &+ I_{2} \left(\frac{\partial \delta \dot{w}_{b}}{\partial x} \frac{\partial \delta \dot{w}_{b}}{\partial x} + \frac{\partial \delta \dot{w}_{b}}{\partial y} \frac{\partial \delta \dot{w}_{b}}{\partial y} \right) \\ &+ K_{2} \left(\frac{\partial \delta \dot{w}_{s}}{\partial x} \frac{\partial \delta \dot{w}_{s}}{\partial x} + \frac{\partial \delta \dot{w}_{s}}{\partial y} \frac{\partial \delta \dot{w}_{s}}{\partial y} \right) \\ &+ J_{2} \left(\frac{\partial \dot{w}_{b}}{\partial x} \frac{\partial \delta \dot{w}_{s}}{\partial x} + \frac{\partial \dot{w}_{s}}{\partial x} \frac{\partial \delta \dot{w}_{s}}{\partial x} + \frac{\partial \dot{w}_{b}}{\partial y} \frac{\partial \delta \dot{w}_{s}}{\partial y} \right) \\ &+ J_{2} \left(\frac{\partial \dot{w}_{b}}{\partial x} \frac{\partial \delta \dot{w}_{s}}{\partial x} + \frac{\partial \dot{w}_{s}}{\partial x} \frac{\partial \delta \dot{w}_{b}}{\partial x} + \frac{\partial \dot{w}_{b}}{\partial y} \frac{\partial \delta \dot{w}_{s}}{\partial y} + \frac{\partial \dot{w}_{s}}{\partial y} \frac{\partial \delta \dot{w}_{s}}{\partial y} \right) \right\} dA \end{split}$$

In above relation, the differentiation with respect to time

is introduced with the dot-superscript; and $(I_0, I_1, J_1, I_2, J_2, K_2)$ are mass inertias as below

$$\{I_0, I_1, J_1, I_2, J_2, K_2\} = \int_{-h/2}^{h/2} \{1, z, f, z^2, zf, f^2\} \rho(z) dz$$
 (24)

Inserting the expressions for δU , δV , and δK from Eqs. (17), (20), and (23) into Eq. (16) and integrating by parts from them, with collecting the coefficients of δu_0 , δv_0 , δw_b , and δw_s equilibrium equations are derived as

$$\delta u_0: \frac{\partial N_x}{\partial x} + \frac{\partial N_{xy}}{\partial y} = I_0 \ddot{u_0} - I_1 \frac{\partial \ddot{w_b}}{\partial x} - J_1 \frac{\partial \ddot{w_s}}{\partial x}$$
 (25)

$$\delta v_0: \frac{\partial N_{xy}}{\partial x} + \frac{\partial N_y}{\partial y} = I_0 \ddot{v_0} - I_1 \frac{\partial \ddot{w_b}}{\partial y} - J_1 \frac{\partial \ddot{w_s}}{\partial y}$$
 (26)

$$\delta w_{b} : \frac{\partial^{2} M_{x}^{b}}{\partial x^{2}} + 2 \frac{\partial^{2} M_{xy}^{b}}{\partial x \partial y} + \frac{\partial^{2} M_{y}^{b}}{\partial y^{2}} - k_{w} w + k_{p} \left(\frac{\partial^{2} w}{\partial x^{2}} + \frac{\partial^{2} w}{\partial y^{2}} \right)$$

$$= I_{0} \left(\ddot{w}_{b} + \ddot{w}_{s} \right) + I_{1} \left(\frac{\partial \ddot{u}_{0}}{\partial x} + \frac{\partial \ddot{v}_{0}}{\partial y} \right) - I_{2} \nabla^{2} \ddot{w}_{b} - J_{2} \nabla^{2} \ddot{w}_{s}$$

$$(27)$$

$$\delta w_{s} : \frac{\partial^{2} M_{x}^{s}}{\partial x^{2}} + 2 \frac{\partial^{2} M_{xy}^{s}}{\partial x \partial y} + \frac{\partial^{2} M_{y}^{s}}{\partial y^{2}} + \frac{\partial Q_{xz}^{s}}{\partial x} + \frac{\partial Q_{yz}^{s}}{\partial y} - k_{w}w
+ k_{P} \left(\frac{\partial^{2} w}{\partial x^{2}} + \frac{\partial^{2} w}{\partial y^{2}}\right) = I_{0} \left(\ddot{w}_{b} + \ddot{w}_{s}\right) + J_{1} \left(\frac{\partial \ddot{u}_{0}}{\partial x} + \frac{\partial \ddot{v}_{0}}{\partial y}\right)
- J_{2} \nabla^{2} \ddot{w}_{b} - K_{2} \nabla^{2} \ddot{w}_{s}$$
(28)

2.4 Equations of motion

Based on the nonlocal strain gradient theory, the constitutive relations of presented higher order nanoplate can be stated as

$$L_{\mu}\begin{bmatrix} \sigma_{x} \\ \sigma_{y} \\ \tau_{xz} \\ \tau_{yz} \\ \tau_{xy} \end{bmatrix} = \begin{bmatrix} C_{11} & C_{12} & 0 & 0 & 0 \\ C_{12} & C_{22} & 0 & 0 & 0 \\ 0 & 0 & C_{44} & 0 & 0 \\ 0 & 0 & 0 & C_{55} & 0 \\ 0 & 0 & 0 & 0 & C_{66} \end{bmatrix} \times L_{I} \begin{bmatrix} \varepsilon_{x} \\ \varepsilon_{y} \\ \gamma_{xz} \\ \gamma_{yz} \\ \gamma_{xy} \end{bmatrix}$$
 (29)

in which $(\sigma_x, \sigma_y, \tau_{yz}, \tau_{xz}, \tau_{xy})$ and $(\varepsilon_x, \varepsilon_y, \gamma_{yz}, \gamma_{xz}, \gamma_{xy})$ denote the stress and strain components, respectively. Integrating Eq. (32), the force strain and the moment-strain of gradient refined FG plate model can be presented as

$$L_{1}\left\{A_{11}\frac{\partial^{2}u_{0}}{\partial x^{2}}+A_{66}\frac{\partial^{2}u_{0}}{\partial y^{2}}+(A_{12}+A_{66})\frac{\partial^{2}v_{0}}{\partial x\,\partial y}-B_{11}\frac{\partial^{3}w_{b}}{\partial x^{3}}\right\} +L_{1}\left\{-(B_{12}+2B_{66})\frac{\partial^{3}w_{b}}{\partial x\,\partial y^{2}}-B_{11}^{s}\frac{\partial^{3}w_{s}}{\partial x^{3}}-(B_{12}^{s}+2B_{66}^{s})\frac{\partial^{3}w_{s}}{\partial x\,\partial y^{2}}\right\} =L_{\mu}\left\{I_{0}\ddot{u}_{0}-I_{1}\frac{\partial \ddot{w}_{b}}{\partial x}-J_{1}\frac{\partial \ddot{w}_{s}}{\partial x}\right\}$$
(30)

$$L_{1}\left\{A_{66}\frac{\partial^{2}v_{0}}{\partial x^{2}}+A_{22}\frac{\partial^{2}v_{0}}{\partial y^{2}}+(A_{12}+A_{66})\frac{\partial^{2}u_{0}}{\partial x\partial y}-B_{22}\frac{\partial^{3}w_{b}}{\partial y^{3}}\right\} +L_{1}\left\{-(B_{12}+2B_{66})\frac{\partial^{3}w_{b}}{\partial x^{2}\partial y}-B_{22}\frac{\partial^{3}w_{s}}{\partial x^{3}}-(B_{12}^{s}+2B_{66}^{s})\frac{\partial^{3}w_{s}}{\partial x^{2}\partial y}\right\}$$

$$=L_{1}\left\{I_{0}\ddot{v_{0}}-I_{1}\frac{\partial \ddot{w}_{b}}{\partial y}-J_{1}\frac{\partial \ddot{w}_{s}}{\partial y}\right\}$$
(31)

$$L_{l} \left\{ B_{11} \frac{\partial^{3} u_{0}}{\partial x^{3}} + (B_{12} + 2B_{66}) \frac{\partial^{3} u_{0}}{\partial x \partial y^{2}} + (B_{12} + 2B_{66}) \frac{\partial^{3} v_{0}}{\partial x^{2} \partial y} \right\}$$

$$+ L_{l} \left\{ + B_{22} \frac{\partial^{3} v_{0}}{\partial y^{3}} - D_{11} \frac{\partial^{4} w_{b}}{\partial x^{4}} - 2(D_{12} + 2D_{66}) \frac{\partial^{4} w_{b}}{\partial x^{2} \partial y^{2}} - D_{22} \frac{\partial^{4} w_{b}}{\partial x^{4}} \right\}$$

$$+ L_{l} \left\{ -D_{11}^{s} \frac{\partial^{4} w_{s}}{\partial x^{4}} - 2(D_{12}^{s} + 2D_{66}^{s}) \frac{\partial^{4} w_{s}}{\partial x^{2} \partial y^{2}} - D_{22}^{s} \frac{\partial^{4} w_{s}}{\partial y^{4}} \right\}$$

$$= L_{\mu} \left\{ k_{w} w - k_{p} \left(\frac{\partial^{2} w}{\partial x^{2}} + \frac{\partial^{2} w}{\partial y^{2}} \right) + I_{0} \left(\ddot{w}_{b} + \ddot{w}_{s}^{s} \right) + I_{1} \left(\frac{\partial \ddot{u}_{0}}{\partial x} + \frac{\partial \ddot{v}_{0}}{\partial y} \right) \right\}$$

$$-I_{2} \nabla^{2} \ddot{w}_{b} - J_{2} \nabla^{2} \ddot{w}_{s}$$

$$\begin{split} & L_{I} \left\{ B_{11}^{s} \frac{\partial^{3} u_{0}}{\partial x^{3}} + (B_{12}^{s} + 2B_{66}^{s}) \frac{\partial^{3} u_{0}}{\partial x \partial y^{2}} + (B_{12}^{s} + 2B_{66}^{s}) \frac{\partial^{3} v_{0}}{\partial x^{2} \partial y} \right\} \\ & + L_{I} \left\{ + B_{22} \frac{\partial^{3} v_{0}}{\partial y^{3}} - D_{11}^{s} \frac{\partial^{4} w_{b}}{\partial x^{4}} A_{55}^{s} \frac{\partial^{2} w_{s}}{\partial x^{2}} + A_{44}^{s} \frac{\partial^{2} w_{s}}{\partial y^{2}} - \right\} \\ & + L_{I} \left\{ 2(D_{12}^{s} + 2D_{66}^{s}) \frac{\partial^{4} w_{b}}{\partial x^{2} \partial y^{2}} - D_{22}^{s} \frac{\partial^{4} w_{b}}{\partial x^{4}} - H_{11}^{s} \frac{\partial^{4} w_{s}}{\partial x^{4}} \right\} \\ & + L_{I} \left\{ -2(H_{12}^{s} + 2H_{66}^{s}) \frac{\partial^{4} w_{s}}{\partial x^{2} \partial y^{2}} - H_{22}^{s} \frac{\partial^{4} w_{s}}{\partial y^{4}} \right\} \\ & = L_{\mu} \left\{ k_{w} w - k_{p} (\frac{\partial^{3} w}{\partial x^{2}} + \frac{\partial^{2} w}{\partial y^{2}}) + I_{0} (\ddot{w}_{b} + \ddot{w}_{s}) J_{1} \left(\frac{\partial \ddot{u}_{0}}{\partial x} + \frac{\partial \ddot{v}_{0}}{\partial y} \right) \right\} \\ & - J_{2} \nabla^{2} \ddot{w}_{b} - K_{2} \nabla^{2} \ddot{w}_{s} \end{split}$$

3. Solution procedure

In order to study wave propagation in FG anisotropic nanoplate, displacement along u_0 , v_0 , w_b and w_s is expressed as

$$u_{0} = A_{1} e^{i(xk_{1}+yk_{2}-\omega t)}$$

$$v_{0} = A_{2} e^{i(xk_{1}+yk_{2}-\omega t)}$$

$$w_{b} = A_{3} e^{i(xk_{1}+yk_{2}-\omega t)}$$

$$w_{s} = A_{4} e^{i(xk_{1}+yk_{2}-\omega t)}$$
(34)

where A_1, A_2, A_3 and A_4 are wave amplitudes; k_1 and k_2 are wave numbers along x, y directions.

$$([K] - \omega^2[M])\{\Delta\} = 0$$
 (35)

where [K] and [M], respectively, are the stiffness matrix and mass matrix; ω is circular frequency. Substituting Eq. (34) into Eqs. (30)-(33), the equations of motion are written in matrix form as follows

$$\left[\left[K \right] - \omega^2 \left[M \right] \right] = 0 \tag{36}$$

To find the wave eigenfrequencies ω , the determinant of above matrix is set to zero. However, phase velocity of waves propagating in a structure is an important parameter in analysis of wave propagation. Phase velocity depends on the obtained wave frequency as well as wave number according to the following relation

$$c = \omega/k \tag{37}$$

in which $k_1=k_2=k$.

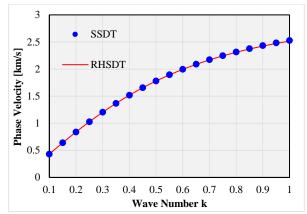
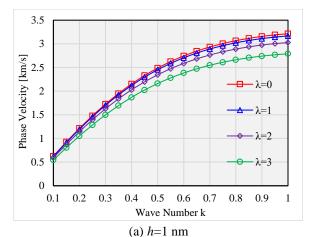


Fig. 2 Comparison of nonlocal strain gradient wave dispersion curves for rectangular nanoplate. $(l = 0.2 \, \text{nm}, \, \mu = 1 \, \text{nm})$



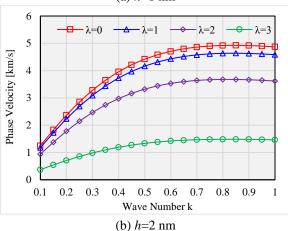
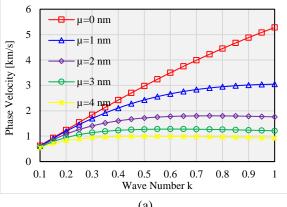


Fig. 3 Influence of exponential factor on the wave dispersion curves of FG anisotropic nanoplate. $(l = 0.2 \, \text{nm}, \, \mu = 1 \, \text{nm})$

4. Numerical results and discussions

After proposing analytical states for nonlocal strain gradient refined model of FG anisotropic nanoplates, selected numerical results are presented in this section.

Firstly, the efficiency of present methodology is verified. Because in accordance with the best authors knowledge there is no study in the open literature on the



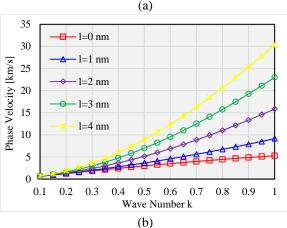


Fig. 4 Nonlocal strain gradient wave dispersion curves of FG anisotropic nanoplate. (a) l=0 nm, (b) μ =0nm

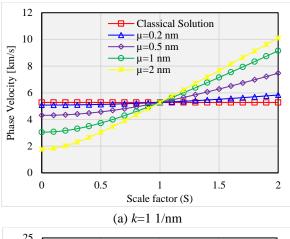
size-dependent wave propagation of FG anisotropic nanoplates. Considering the nonlocal strain gradient effects the wave propagation of nanoplates made of an isotropic material with no associated to boundary conditions (bulk waves) are illustrated in Fig. 2 and compared with those obtained by Karami *et al.* (2018h) using second order shear deformation theory (SSDT). An excellent agreement is found between two types of modeling process which confirms the validity of the given preceding numerical results.

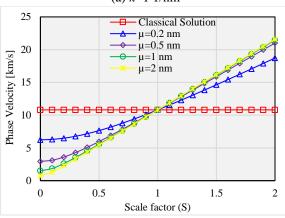
4.1 Thickness of nanoplates and exponential factor

Fig. 3 displays the nonlocal strain gradient phase velocity curves of FG anisotropic nanoplates corresponding to various values of wave numbers. Also, varying of thicknesses in nanoplates and exponential factor effects are shown. It is found that the with increasing exponential factor and thickness of nanoplate, the phase velocity of FG anisotropic nanoplate will decrease and increase respectively. As a consequence, the exponential factor effect on the phase velocity of FG anisotropic nanoplate is more in the high value of the thickness of nanoplate.

4.2 Nonlocal constant and strain gradient length scale parameter

Fig. 4 depicts the nonlocal strain gradient phase velocity





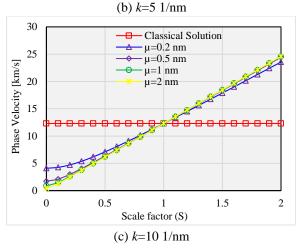


Fig. 5 Variation of phase velocities of FG anisotropic nanoplate with respect to scale factor and nonlocal parameter

curves of FG anisotropic nanoplates with different small scale parameters. It is seen that the softening-stiffness effect of nonlocality causes to reduce the phase velocity, while the hardening-stiffness of strain gradient size dependency leads to enhance them. In addition, it can be observed that both types of size effect have more influence on the phase velocity in higher values of wave number.

Nonlocality effects are shown in Fig. 5 for variations of phase velocities in FG anisotropic nanoplate as a function of scale factor (S), where $S = \frac{l}{u}$.

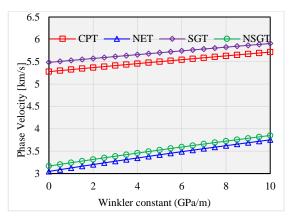


Fig. 6 Phase velocity based on Winkler constant and in different theory

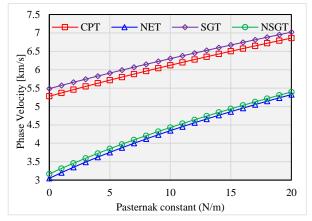


Fig. 7 Phase velocity based on Pasternak constant and in different theory

As we see in figure, when the scale factor (S) is less than unity, the nanoplate provides softer response and the size-dependent phase velocities are smaller than those from classical model. Also, it can be concluded that for values of nonlocal parameter (μ) smaller than gradient parameter (l), the results achieved from present theory are larger than those from the classical model. Besides, more changes can be seen in lower value of wave number with the variation of the scale factor (S). It is mentioned that by neglecting the scale factor (S), the present results are the same as the results from Eringen's nonlocal theory.

4.3 Winkler constant

Fig. 6 demonstrates the effect of Winkler constant of FG anisotropic nanoplate on the phase velocity in different theories (CPT; Nonlocal Elasticity Theory (NET); Strain Gradient Theory (SGT); Nonlocal Strain Gradient Theory (NSGT)). Increased Winkler constants leads to increased nanoplate rigidity, which in turn leads to increase the phase velocity. It can be seen that the phase velocity has the highest and lowest values for the strain gradient theory and the nonlocal elasticity theory, respectively. It means that with increasing the gradient parameter, the rigidity of FG anisotropic nanoplate will increase.

Table 1 Frequency of rectangular FG anisotropic nanoplate

k (1/nm)	λ	μ (nm)	l (nm)	h= 1 nm				h= 2 nm			
				T_1	T_2	T_3	T_4	T_1	T_2	T_3	T_4
1	0	0	0	5.6111	2.3478	2.3478	0.8526	3.7096	2.3478	2.3478	1.2920
			1.5	13.1592	5.5060	5.5060	1.9995	8.6997	5.5060	5.5060	3.0301
			3	24.4582	10.2337	10.2337	3.7164	16.1696	10.2337	10.2337	5.6319
		1	0	3.2396	1.3555	1.3555	0.4922	2.1417	1.3555	1.3555	0.7460
			1.5	7.5975	3.1789	3.1789	1.1544	5.0228	3.1789	3.1789	1.7494
			3	14.1210	5.9084	5.9084	2.1457	9.3355	5.9084	5.9084	3.2516
		2	0	1.8704	0.7826	0.7826	0.2842	1.2365	0.7826	0.7826	0.4307
			1.5	4.3864	1.8353	1.8353	0.6665	2.8999	1.8353	1.8353	1.0100
			3	8.1527	3.4112	3.4112	1.2388	5.3899	3.4112	3.4112	1.8773
	1	0	0	5.6989	2.3893	2.3893	0.8402	4.1868	2.4418	2.4418	1.2161
	•	Ü	1.5	13.3651	5.6034	5.6034	1.9703	9.8188	5.7264	5.7264	2.8520
			3	24.8409	10.4147	10.4147	3.6622	18.2496	10.6434	10.6434	5.3008
		1			1.37947					1.4097	0.7021
		1	0	3.2903		1.37947	0.4851	2.4172	1.4097		
			1.5	7.7163	3.23514	3.23514	1.1376	5.6689	3.3062	3.3062	1.6466
			3	14.3419	6.01295	6.01295	2.1143	10.5364	6.1449	6.1449	3.0604
		2	0	1.8996	0.7964	0.7964	0.2801	1.3956	0.8139	0.8139	0.4054
			1.5	4.4550	1.8678	1.8678	0.6568	3.2729	1.9088	1.9088	0.9507
			3	8.2803	3.4716	3.4716	1.2207	6.0832	3.5478	3.5478	1.7669
	2	0	0	5.9668	2.5135	2.5135	0.8031	5.4980	2.8023	2.8023	0.9597
			1.5	13.9933	5.8948	5.8948	1.8835	12.8939	6.5720	6.5720	2.2506
			3	26.0085	10.9562	10.9562	3.5007	23.9651	12.2151	12.2151	4.1831
		1	0	3.4449	1.4512	1.4512	0.4637	3.1743	1.6179	1.6179	0.5541
			1.5	8.0790	3.4033	3.4033	1.0874	7.4443	3.7944	3.7944	1.2994
			3	15.0160	6.3256	6.3256	2.0211	13.8363	7.0524	7.0524	2.4151
		2	0	1.9889	0.8378	0.8378	0.2677	1.8327	0.9341	0.9341	0.3199
			1.5	4.6644	1.9649	1.9649	0.6278	4.2980	2.1907	2.1907	0.7502
			3	8.6695	3.6521	3.6521	1.1669	7.9884	4.0717	4.0717	1.3944
2	0	0	0	7.4191	4.6955	4.6955	2.5841	5.8536	4.6955	4.6955	3.2988
			1.5	32.3392	20.4674	20.4674	11.2638	25.5151	20.4674	20.4674	14.3791
			3	63.3890	40.1187	40.1187	22.0786	50.0129	40.1187	40.1187	28.1848
		1	0	2.4730	1.5652	1.5652	0.8614	1.9512	1.5652	1.5652	1.0996
			1.5	10.7797	6.8225	6.8225	3.7546	8.5050	6.8225	6.8225	4.7930
			3	21.1297	13.3729	13.3729	7.3595	16.6710	13.3729	13.3729	9.3949
		2	0	1.2915	0.8174	0.8174	0.4498	1.0190	0.8174	0.8174	0.5742
			1.5	5.6295	3.5630	3.5630	1.9608	4.4416	3.5629	3.5629	2.5031
			3	11.0346	6.9838	6.9838	3.8434	8.7061	6.9838	6.9838	4.9063
	1	0	0	7.6682	4.7373	4.7373	2.5466	7.2849	4.6515	4.6515	3.1131
			1.5	33.4247	20.6493	20.6493	11.1005	31.7541	20.2752	20.2752	13.5697
			3	65.5168	40.4754	40.4754	21.7584	62.2421	39.7420	39.7420	26.5983
		1	0	2.5561	1.5791	1.5791	0.8489	2.4283	1.5505	1.5505	1.0377
			1.5	11.1416	6.8831	6.8831	3.7002	10.5847	6.7584	6.7584	4.5232
			3	21.8389	13.4918	13.4918	7.2528	20.7474	13.2473	13.2473	4.3232 8.8661
		2									
		2	0	1.3349	0.8247	0.8247	0.4433	1.2681	0.8097	0.8097	0.5419
			1.5	5.8185	3.5946	3.5946	1.9324	5.5277	3.5295	3.5295	2.3622
			3	11.4050	7.0459	7.0459	3.7877	10.8350	6.9182	6.9182	4.6302
	2	0	0	8.3735	4.8835	4.8835	2.4322	10.1712	5.6995	4.9249	2.3575

Table 1 Continued

	1.5	36.4993	21.2867	21.2867	10.6015	44.3353	24.8435	21.4673	10.2762
	3	71.5433	41.7247	41.7247	20.7803	86.9030	48.6964	42.0786	20.1427
1	0	2.7912	1.6278	1.6278	0.8107	3.3904	1.8998	1.6416	0.7858
	1.5	12.1664	7.0956	7.0956	3.5338	14.7784	8.2812	7.1558	3.4254
	3	23.8478	13.9082	13.9082	6.9268	28.9677	16.2321	14.0262	6.7142
2	0	1.4576	0.8501	0.8501	0.4234	1.7706	0.9922	0.8573	0.4104
	1.5	6.3537	3.7055	3.7055	1.8455	7.7178	4.3247	3.7370	1.7889
	3	12.4541	7.2633	7.2633	3.6174	15.1279	8.4770	7.3249	3.5064

4.4 Pasternak constant

Fig. 7 demonstrates the effects of Pasternak constant of FG anisotropic nanoplate on the phase velocity in different theories. An increase in Pasternak constant leads to an increase in the phase velocity. In this case, increasing in Pasternak constant and gradient parameter is accompanied by an increased in the rigidity of the FG anisotropic nanoplate while the nonlocal theory has the behavior contrary to them.

4.5 Four different modes for wave analysis

One of the aims of this study is to provide the wave behavior of rectangular FG anisotropic nanoplates with respect to nonlocal parameter (μ) and strain gradient length scale parameter (l). So, as a benchmark table, wave characteristics of FG anisotropic nanoplate for the wave frequency for different nonlocal parameter (μ) , strain gradient length scale (l), wave number, exponential factor (λ) and thickness of the plate (h) has been tabulated in Table 1. To provide results, results are explored for four pairs of distinct dispersion modes, (T_1, T_2, T_3, T_4) . As can be seen, the wave frequency of FG anisotropic nanoplate rises with the increment of the strain gradient length scale (l) and reduces with the increment of the nonlocal parameter (μ) .

5. Conclusions

The objective of present work was to analyze the size-dependent wave propagation behavior of rectangular FG anisotropic nanoplates rested on elastic foundation including both of hardening-stiffness and softening-stiffness size effects. To this end, the nonlocal strain gradient theory of elasticity was incorporated to the RPT the governing equations of wave motion was obtained employing the Hamiltonian principle. After that, an analytical technique was used to solve the governing equations for wave propagation of FG anisotropic nanoplates as a function of wave number.

It was revealed that the softening-stiffness influence of nonlocality causes to reduce the phase velocity, while the hardening-stiffness of strain gradient size dependency leads to enhance them. Moreover, it was observed that the wave number cause enhances the phase velocity of FG anisotropic nanoplate. It was revealed that the softening-

stiffness effect of nonlocality causes to reduce the phase velocity, while the hardening-stiffness of strain gradient size dependency leads to enhance them. Moreover, it was observed that the wave number cause enhances the phase velocity of FG anisotropic nanoplate. Additionally, it was seen that both of the electric foundation parameters cause to increase of the phase velocity of FG anisotropic nanoplates. Furthermore, increasing the exponential factor induces a reduction effect on the obtained results for phase velocity. In addition, it was demonstrated that the thickness of plate had the more important role on the wave behavior of FG anisotropic nanoplate.

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