

# Response of orthotropic Kelvin modeling for single-walled carbon nanotubes: Frequency analysis

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**Abstract.** In this paper, modified Kelvin's model has been used to analyze the orthotropic vibration frequencies of single walled carbon nanotubes with clamped-clamped and clamped-free boundary conditions. For this system the governing equation is developed with wave propagation approach. Armchair, zigzag and chiral structures are considered for the vibrational analysis to investigate the effect of different modes, in-plane rigidity and mass density per unit lateral area. Throughout the computations, on decreasing the length-to-diameter ratios, the frequencies of said structure increases. In addition, by increasing three different value of in-plane rigidity resulting frequencies also increase and frequencies decrease on increasing mass density per unit lateral area. The results generated using computer software MATLAB to furnish the evidence regarding applicability of present model and also verified by available published literature.

**Keywords:** CNTs; wave propagation approach; Kelvin Model; bending rigidity

## 1. Introduction

CNTs have a variety of applications because of their distinctive molecular structure and show unique electronic and mechanical properties because of their curvature. In addition, they are utilized in different fields such as bioengineering, tissue engineering, computer engineering, optics, energy and environmental systems. Here, a succinct review has been provided regarding different applications of CNTs. Their enormous versatility has made them being coined as a solution looking for problems. Natuski *et al.* (2006) conducted the vibration of nested CNTs in elastic matrix. Flugge shell theory had been engaged to establish administrative shell equations while proposed method was wave propagation. Wang and Liew (2007) studied nano- and micro-tubes using EBM and TBM. The results showed that to indicate the importance of applied theory of beams, the shear effect is evident for the carbon nanotubes. Furthermore, vibrations of nanotubes have been modeled by atomistic/continuum models. actuators, sensors and sonication.

The fundamental frequencies of CNTs have higher values at lower diameter and showed that the chirality have no momentous influence on frequencies of nanotubes. The vacancies are produced due to holes in the graphite structure and due to chemical and theoretical physics, the

defects in CNTs reduce the Young's modulus (Kotakoski *et al.* 2006). Usuki and Yogo (2009) formed beam equations again based on Flugge shell theory, they concluded that if nonlocality and refined model are ignored then the generalized beam theory and Flugge theory produce alike results. Owing to this, Gibson *et al.* (2007) presented a coherent review about the simulations, and modeling of vibrating nanotubes. Akgöz and Civalek (2012) examined with the bending analysis of single-walled carbon nanotubes (CNT) based on modified couple stress and strain gradient elasticity theories and Euler–Bernoulli beam theory. The size effect is taken into consideration using the modified couple stress and strain gradient elasticity theories. Kulathunga *et al.* (2009) widely explored the classical shell theory to compute the strain of SWCNTs. It has been calculated for the first time from this formula in the absence of length-diameter ratio, the critical buckling strain can be calculated. Mehar and Panda (2016a, b, 2018a) computed the vibration behavior, bending and dynamic response of FG reinforced CNT using shear deformation theory and finite element method. For the sake of generality, the mathematical model was presented with the mixture of Green Lagrange method. The convergence of these methodologies has been checked for the variety of results. The composite plates with different graded was investigated with isotropic and core phase. Avcar (2015) presented the separate and combined effects of rotary inertia, shear deformation and material non-homogeneity (MNH) on the values of natural frequencies of the simply supported beam. MNH is characterized considering the parabolic variations of Young's modulus and density along the thickness direction of the beam, while the value of

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Poisson's ratio is assumed to remain constant. The results are verified with the molecular dynamics (MD) simulation based on compass force field. This improved formula shows a significant error at the smaller diameters and higher aspect ratios. Mehar *et al.* (2017a, b, c, d) studied the frequency response of FG CNT and reinforced CNT using the simple deformation theory, finite element modeling, Mori-Tanaka scheme. They investigated a new frequency phenomenon with the combination of Lagrange strain, Green-Lagrange, for double curved and curved panel of FG and reinforced FG CNT. The characteristics of sandwich and grades CNT was found with labeling the temperature environ. The thermoelastic frequency of single shallow panel was determined using Mori-Tanaka formulations. The research of these authors has opened a new frequency spectrum for other material researchers. Chavan and Lal (2017) presented the bending characteristic of single wall carbon nanotube reinforced functionally graded composite (SWCNTRC-FG) plates. The finite element implementation of bending analysis of laminated composite plate via well-established higher order shear deformation theory (HSDT). Selim (2010) performed vibrational behavior of SWCNT with compression stresses using multilayers. The motion equation was derived for the vibration analysis of SWCNTs using wave propagation. However, the outcomes of this research noted for various structures and the overall electronic and structural properties. Civalek *et al.* (2018) examined the present paper, rotary inertia and small size effects on the free vibration response of single-walled carbon nanotubes. The equations in motion and associated boundary conditions are obtained by using Hamilton's principle on the basis of modified couple stress and Rayleigh beam theories. Mehar and Panda (2018b) investigated the curved shell and CNT vibration with thermal environment using higher order deformation theory. This CNT was mixed with different configurations of the layers. The results have been verified with the earlier investigations. Ebrahimi and Habibi (2017) studied the nonlinear response of laminated functionally graded carbon nanotube reinforced composite (FG-CNTRC) plate under low-velocity impact based on the Eshelby-Mori-Tanaka approach in thermal conditions. The governing equations are derived based on higher-order shear deformation plate theory (HSDT) under von Kármán geometrical nonlinearity assumptions. Azrar *et al.* (2011) have explained the effect of vibration of these tubes on the length scale of many end conditions using TBM and nonlocal elasticity theory by solving the generalized nonlinear algebraic equations; higher and lower frequencies associated with eigen-modes are obtained. Emdadi *et al.* (2019) investigated the free vibration analysis of annular sandwich plates with various functionally graded (FG) porous cores and carbon nanotubes reinforced composite (CNTRC) facesheets based on modified couple stress theory (MCST) and first order shear deformation theories (FSDT). Mehar and Panda (2018c) investigated numerically the deflection behavior of carbon nanotube reinforced composite plate is using the finite element method and the result accuracy is established via threepoint experimental bending test data. Fazelzadeh and Ghavanloo (2012a, b) developed a nonlocal anisotropic

elastic shell model to study the effect of small scale on shell-like vibration of single-walled carbon nanotubes (SWCNTs) with arbitrary chirality. Anisotropic elastic shell model is reformulated using the nonlocal differential constitutive relations of Eringen. Single-walled carbon nanotubes (SWCNTs) exhibit remarkable chirality-dependent mechanical phenomena. In present paper, an anisotropic elastic shell model is developed to study the vibration characteristics of chiral SWCNTs.

Rouhi *et al.* (2013) executed the axial buckling of double-walled CNT subject to various layer-wise conditions by using Rayleigh-Ritz based upon nonlocal Flügge shell theory. Their study showed that the number of different layer-wise boundary conditions dominates the choice of values for nonlocal parameter. Mehar and Panda (2018d) developed a general mathematical model for the evaluation of the theoretical flexural responses of the functionally graded carbon nanotube reinforced composite doubly curved shell panel using higher-order shear deformation theory with thermal load. Hussain and Naeem (2017) examined the frequencies of armchair tubes using Flügge's shell model. The effect of length and thickness-to-radius ratios against fundamental natural frequency with different indices of armchair tube was investigated. Mohammadimehr and Alimirzaei (2016) analyzed the nonlinear static and free vibration analysis of Euler-Bernoulli composite beam model reinforced by functionally graded single-walled carbon nanotubes (FG-SWCNTs) with initial geometrical imperfection under uniformly distributed load using finite element method (FEM) is investigated. The governing equations of equilibrium are derived. Rafiee and Moghadam (2012) employed a 3D-FEM with reinforced polymer CNT. The shear deformation of CNTs is implicated as using TBM and results are compared with earlier simulations. The behavior of reinforced polymer CNT is evaluated both in axial and transverse direction subject to high strain loading and impact of buckling are stimulated at micro-scale. Sofiyev and Avcar (2010) studied the stability of cylindrical shells that composed of ceramic, FGM, and metal layers subjected to axial load and resting on Winkler-Pasternak foundations. Hussain *et al.* (2017) demonstrated an overview of Donnell theory for the frequency characteristics of two types of SWCNTs. Fundamental frequencies with different parameters have been investigated with wave propagation approach. Mehar and Panda (2018e) reported the nonlinear finite solutions of the nonlinear flexural strength and stress behaviour of nano sandwich graded structural shell panel under the combined thermomechanical loading. Ansari and Rouhi (2013) summarized the effect of small scale, geometrical parameter and layer-wise end conditions of double-walled CNT by adopting Flügge shell model (FSM). They depicted that the continuum model considering the nonlocal effect compels the short double-walled CNT more flexible. Mehar *et al.* (2018a) evaluated the frequency behavior of nanoplate structure using FEM including the nonlocal theory of elasticity. Computer generated results are created by using the software first time robustly to check the vibration of nanoplate. The efficiency was checked by comparing the results of available data. Hussain and Naeem (2018a) used

Donnell's shell model to calculate the dimensionless frequencies for two types of single-walled carbon nanotubes. The frequency influence was observed with different parameters. Tohidi *et al.* (2018) considered an analysis into the nonlinear forced vibration of a micro cylindrical shell reinforced by carbon nanotubes (CNTs) with considering agglomeration effects. The structure is subjected to magnetic field and transverse harmonic mechanical load. In addition, Fereidoon *et al.* (2013) executed the 3D finite element method to find the modal analysis of reinforced polymer CNT with two type of edge condition. The frequencies obtained through finite element method showed a reasonable increment on comparing with neat polymer. Moghadam *et al.* (2014) scrutinized the frequencies of SWCNTs using molecular mechanics approach with various parameters. The frequency outcomes revealed the frequencies increases or decreases with the orientation of carbon-carbon bond.

Mehar *et al.* (2018b) studied the bending responses of nanotube-reinforced curved sandwich shell panel structure under the influence of the thermomechanical loading. Further, the temperature dependent material properties of the sandwich structure are assumed to evaluate the exact responses. Fatahi-Vajari *et al.* (2019) studied the vibration of single-walled carbon nanotubes based on Galerkin's and homotopy method. This work analyses the nonlinear coupled axial-torsional vibration of single-walled carbon nanotubes (SWCNTs) based on numerical methods. Two-second order partial differential equations that govern the nonlinear coupled axial-torsional vibration for such nanotube are derived. Madaini *et al.* (2016) studied the vibration of FG-CNT-reinforced piezoelectric cylindrical shells utilizing the method of differential quadrature. The mixture rule of four distribution types was used in the thickness direction. Asghar *et al.* (2019) conducted the vibration of nonlocal effect for double-walled carbon nanotubes using wave propagation approach. Many material parameters are varied for the exact frequencies of many indices of double-walled carbon nanotubes. Karami *et al.* (2017) conducted the analysis analysis of FG nanoplates, spherical nano particles and inisotropic nanoparticles using quasi-3D model, 3D elasticity and nonlocal theory. They investigated the dispersion analysis of FG nanoplates nonlocal strain gradient effect and triaxial magnetic effect. Avcar (2019) conducted the influence of various volume fraction laws using the classical theory for FG beam. By varying the volume fraction laws, the span-to-depth ratio versus frequencies was examined in detail. The frequencies for span to depth ratio with varying volume fraction index were examined in detail. Shahrma *et al.* (2019) studied the functionally graded material using sigmoid law distribution under hygrothermal effect. The Eigen frequencies are investigated in detail. Frequency spectra for aspect ratios have been depicted according to various edge conditions. Semmah *et al.* (2019) studied the buckling analysis of zigzag single walled boron nitride based on Winkler foundation. Effect of different nonlocal parameter was investigated with closed form solution. Recently Hussain and Naeem (2019a, b, c, d, 2020) and performed the vibration of SWCNTs based on wave propagation approach

and Galerkin's method. They investigated many physical parameters for the rotating and non-rotating vibrations of armchair, zigzag and chiral indices. Moreover, the mass density effect of single-walled carbon nanotubes with in-plane rigidity have been calculated for zigzag and chiral indice.

The particular motivation of present work is to investigate the vibration characteristics of armchair, zigzag and chiral SWCNT by means of Kelvin's model based on wave propagation approach. The suggested method to investigate the solution of fundamental eigen relations is wave propagation approach (WPA), which is a well-known and efficient technique to develop the fundamental frequency equations. There is no evidence found in the literature regarding our work. The specific influence of four different end supports based on Kelvin's model such as clamped-clamped (C-C), and clamped-free supported (C-F) is examined in detail. Many material researchers calculated the frequency of CNTs using different techniques, for example, classical molecular dynamics (Han *et al.* 1997), Rayleigh-Ritz method (Swaddiwudhipong *et al.* 1995), continuum models (Duan *et al.* 2007), Galerkin's method (Elishakoff and Pentaras 2009), Raman spectroscopy (Jorio *et al.* 2001) deformation theory (Mehar *et al.* 2016), finite element method (Mehar *et al.* 2018c), Green-Lagrange strain field (Mehar *et al.* 2018c) and Multiscale modeling approach (Mehar and Panda 2019, Mehar *et al.* 2019). Also, the modified Kelvin's model based on the wave propagation approach for estimating orthotropic fundamental natural frequency has been developed to converge more quickly than other methods and models. A large use of tube structures in practical applications makes their theoretical analysis an important field of structural dynamics. Since a tube problem is a physical one, so their vibrational behaviors are distorted by variations of physical and material parameters. Effects of different parameters for in-plane rigidity (IPR) and mass density per unit lateral area (MDA) on the fundamental natural frequencies versus ratio of length-to-diameter have been investigated. In this paper, utilizing the Kelvin's model for the vibration of orthotropic carbon nanotubes provides a governing equation using wave propagation approach with C-C and C-F conditions. The effect of frequencies against length-to-diameter ratio/ aspect ratio by varying the in-plane rigidity mass per unit lateral area is studied. It has been shown that on enhancing bending rigidity for SWCNTs the frequency curves increase. It is observed that the outcomes of frequency response of armchair with prescribed boundary conditions are lower than that of zigzag frequency values and the outcomes of frequency response of chiral with prescribed boundary conditions are lower than that of zigzag frequency values. The presented orthotropic vibration modeling and analysis of carbon nanotubes may be helpful especially in applications such as oscillators and in non-destructive testing.

## 2. Materials and methods

When a graphene sheet is rolled with its hexagonal cells, the structure can be conceptualized as single-walled carbon

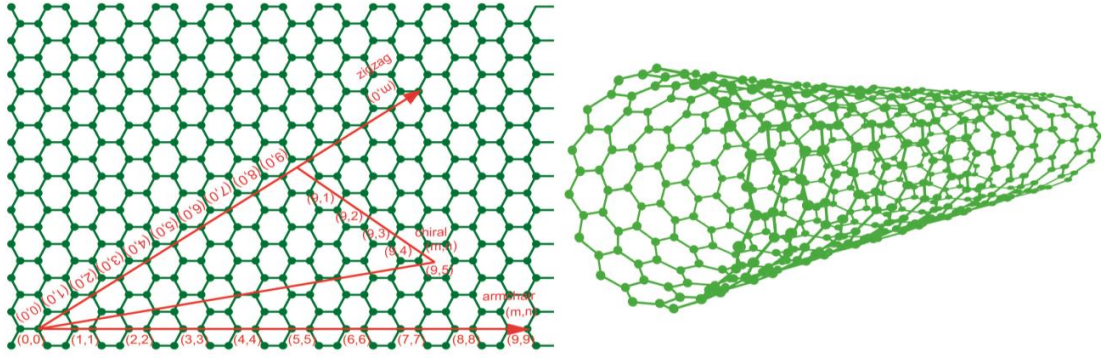


Fig. 1 Hexagonally description of armchair, zigzag and chiral SWCNTs on the graphene sheet

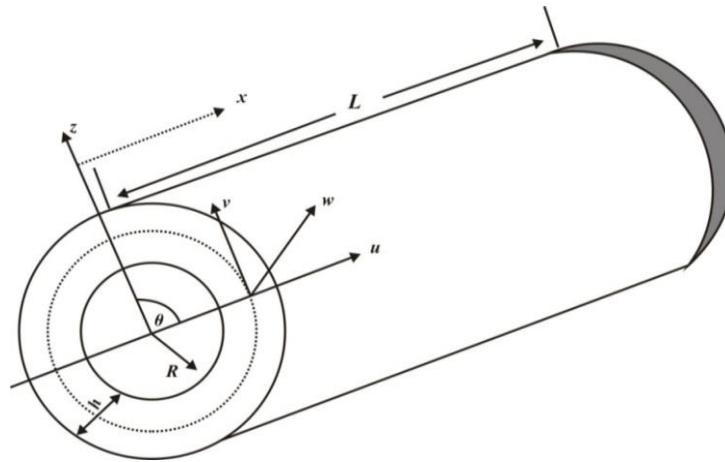


Fig. 2 Geometry of SWCNTs

nanotubes and its circumference and quantum properties depend upon the chirality and diameter described as a pair of  $(n, m)$ . The indices pair occur during the rolling of tube Fig. 1 shows the schema of the pair indices as  $(m, n)$  which occurs on rolling of the tube and this pair of in indices formed as armchair, zigzag and chiral, if  $m = n$ ,  $n = 0$ , if  $m \neq n$ , respectively. The geometry of SWCNTs is shown in Fig. 2.

### 2.1 Orthotropic Kelvin-like model

Over the past several years vibration of tube/shell and plate structures of various configurations and boundary conditions have been extensively studied (Hussain *et al.* 2018a, b, c, 2019a, b, 2020a, b, Hussain and Naeem 2018b, Asghar *et al.* 2020, Taj *et al.* 2020). The geometrical relations are given by Flugge's shell theory (Flugge 1973, Zou and Foster 1995, Paliwal *et al.* 1995). The coordinate system  $x, y$  and  $z$  are axial, circumferential and radial coordinates respectively whose dimensionless coordinates are  $\alpha = x/R$ ,  $\beta = y/R$  and  $\gamma = z/R$  and  $\nabla^2$  is the Laplace operator.

Along  $\alpha$ ,  $\beta$  and  $\gamma$  directions, the displacement of middle surface is  $u, v$  and  $w$ , respectively

$$\varepsilon_\alpha = \frac{1}{R} \left( \frac{\partial u}{\partial \alpha} - \gamma \frac{\partial^2 w}{\partial \alpha^2} \right) \quad (1)$$

$$\varepsilon_\beta = \frac{1}{R} \left( \frac{\partial v}{\partial \beta} + w \right) - \frac{\gamma}{R(1+\gamma)} \left( \frac{\partial^2 w}{\partial \beta^2} + w \right) \quad (2)$$

$$\varepsilon_{\alpha\beta} = \frac{\gamma}{R(1+\gamma)} \left[ \frac{\partial u}{\partial \beta} + \frac{\partial v}{\partial \alpha} + 2\gamma \left( \frac{\partial v}{\partial \alpha} - \frac{\partial^2 w}{\partial \alpha \partial \beta} \right) + \gamma^2 \left( \frac{\partial v}{\partial \alpha} - \frac{\partial^2 w}{\partial \alpha \partial \beta} \right) \right] \quad (3)$$

The stress-strain relationships in dimensionless coordinates is as under (Gao and An 2010)

$$\sigma_\alpha - \nabla^2 \sigma_\alpha = E_1 (\varepsilon_\alpha + \mu_1 \varepsilon_\beta) / (1 - \mu_1 \mu_2) \quad (4)$$

$$\sigma_\beta - \nabla^2 \sigma_\beta = E_2 (\varepsilon_\beta + \mu_2 \varepsilon_\alpha) / (1 - \mu_1 \mu_2) \quad (5)$$

$$\tau_{\alpha\beta} - \nabla^2 \tau_{\alpha\beta} = G \varepsilon_{\alpha\beta} \quad (6)$$

where  $\sigma_\alpha$ ,  $\sigma_\beta$  and  $\tau_{\alpha\beta}$  are normal and shear stresses, and  $\varepsilon_\alpha$ ,  $\varepsilon_\beta$  and  $\varepsilon_{\alpha\beta}$  are respective strains;  $E_1$  and  $E_2$  are moduli of elasticity; Poisson's ratios in the directions of  $\alpha$  and  $\beta$  are  $\mu_2$  and  $\mu_1$  respectively.  $G$  is modulus of rigidity or shear modulus. Also, we have  $E_1 \mu_1 = E_2 \mu_2$  and  $\nabla^2 = (\partial^2 / \partial \alpha^2 + \partial^2 / \partial \beta^2) / R^2$  which is the Laplace operator in dimensionless coordinates. The element of tube in our coordinates is shown in Fig. 3, where  $(N, S, Q)$  are

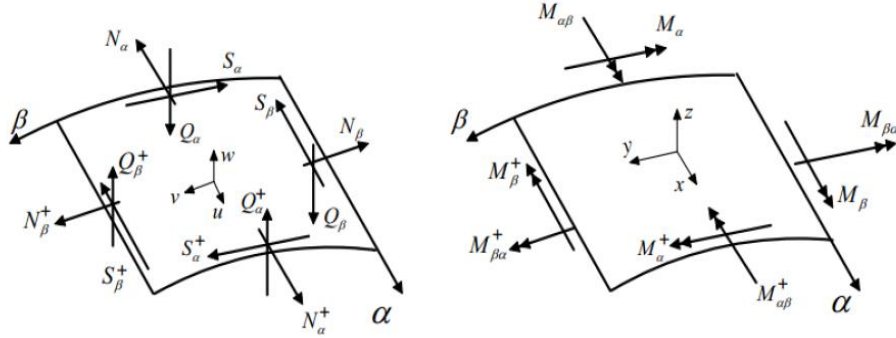


Fig. 3 Resolution of components of stress and moments of the middle surface of CNTs

the stress resultants and ( $M$ ) is the moment. The thermal expansion causes pre-stress, which is neglected due to the reference temperature. We arrive at the dynamic equilibrium equations

$$\begin{cases} \frac{\partial N_\alpha}{\partial \alpha} + \frac{\partial S_\beta}{\partial \beta} + \kappa = \rho h R \frac{\partial^2 u}{\partial t^2} \\ \frac{\partial N_\beta}{\partial \beta} + \frac{\partial S_\alpha}{\partial \alpha} + Q_\beta = \rho h R \frac{\partial^2 v}{\partial t^2} \\ \frac{\partial Q_\alpha}{\partial \alpha} + \frac{\partial Q_\beta}{\partial \beta} + N_\beta = \rho h R \frac{\partial^2 w}{\partial t^2} \end{cases} \quad (7)$$

$$\begin{cases} \frac{\partial M_{\alpha\beta}}{\partial \alpha} + \frac{\partial M_\beta}{\partial \beta} - R Q_\beta = 0 \\ \frac{\partial M_{\beta\alpha}}{\partial \beta} + \frac{\partial M_\alpha}{\partial \alpha} - R Q_\alpha = 0 \end{cases} \quad (8)$$

where  $\rho$  is the mass density.

The resultants ( $N, S, Q$ ) are derived from above set of integral equations using the stress components.

$$(1 - \nabla^2) \begin{bmatrix} N_\alpha, S_\alpha \\ M_\alpha, M_{\alpha\beta} \end{bmatrix} = \int_{-\frac{h}{2}}^{\frac{h}{2}} \begin{bmatrix} \sigma_\alpha, \tau_{\alpha\beta} \\ z \sigma_\alpha, z \tau_{\alpha\beta} \end{bmatrix} \left(1 + \frac{z}{R}\right) dz \quad (9)$$

$$(1 - \nabla^2) \begin{bmatrix} N_\beta, S_\beta \\ M_\beta, M_{\beta\alpha} \end{bmatrix} = \int_{-\frac{h}{2}}^{\frac{h}{2}} \begin{bmatrix} \sigma_\beta, \tau_{\beta\alpha} \\ z \sigma_\beta, z \tau_{\beta\alpha} \end{bmatrix} dz \quad (10)$$

$$(1 - \nabla^2)(Q_\alpha, Q_\beta) = \int_{-\frac{h}{2}}^{\frac{h}{2}} [\tau_{\alpha z}, \tau_{\beta z}] dz \quad (11)$$

where  $h$  is thickness of the shell. Above equations result in

$$N_\alpha - \nabla^2 N_\alpha = \frac{K}{R} \left[ \frac{\partial u}{\partial \alpha} + \mu_1 \left( \frac{\partial v}{\partial \beta} + w \right) - c^2 \frac{\partial^2 w}{\partial \alpha^2} \right] \quad (12)$$

$$N_\beta - \nabla^2 N_\beta = \frac{K k_1}{R} \left[ \frac{\partial v}{\partial \beta} + \mu_2 \frac{\partial u}{\partial \alpha} + w + c^2 \left( \frac{\partial^2 w}{\partial \beta^2} + w \right) \right] \quad (13)$$

$$S_\alpha - \nabla^2 S_\alpha = \frac{K k_2}{R} \left[ \frac{\partial u}{\partial \beta} + \frac{\partial v}{\partial \alpha} - c^2 \left( \frac{\partial^2 w}{\partial \alpha \partial \beta} - \frac{\partial v}{\partial \alpha} \right) \right] \quad (14)$$

$$S_\beta - \nabla^2 S_\beta = \frac{K k_2}{R} \left[ \frac{\partial u}{\partial \beta} + \frac{\partial v}{\partial \alpha} + c^2 \left( \frac{\partial^2 w}{\partial \alpha \partial \beta} + \frac{\partial v}{\partial \alpha} \right) \right] \quad (15)$$

$$M_\alpha - \nabla^2 M_\alpha = -K c^2 \left[ \frac{\partial u}{\partial \alpha} + \mu_1 \frac{\partial v}{\partial \beta} - \left( \frac{\partial^2 w}{\partial \alpha^2} + \mu_1 \frac{\partial^2 w}{\partial \beta^2} \right) \right] \quad (16)$$

$$M_\beta - \nabla^2 M_\beta = K k_1 c^2 \left( \frac{\partial^2 w}{\partial \beta^2} + w + \mu_2 \frac{\partial^2 w}{\partial \alpha^2} \right) \quad (17)$$

$$M_{\alpha\beta} - \nabla^2 M_{\alpha\beta} = 2K k_2 c^2 \left( \frac{\partial v}{\partial \alpha} - \frac{\partial^2 w}{\partial \alpha \partial \beta} \right) \quad (18)$$

$$M_{\beta\alpha} - \nabla^2 M_{\beta\alpha} = K k_2 c^2 \left( \frac{\partial u}{\partial \beta} - \frac{\partial v}{\partial \alpha} + 2 \frac{\partial^2 w}{\partial \alpha \partial \beta} \right) \quad (19)$$

$$Q_\alpha - \nabla^2 Q_\alpha = \frac{K c^2}{R} \left[ \frac{\partial^2 u}{\partial \alpha^2} - k_2 \frac{\partial^2 u}{\partial \beta^2} + (k_2 + \mu_1) \frac{\partial^2 v}{\partial \alpha \partial \beta} - \left[ \frac{\partial^3 w}{\partial \alpha^3} - (2k_2 + \mu_1) \frac{\partial^3 w}{\partial \alpha \partial \beta^2} \right] \right] \quad (20)$$

$$Q_\beta - \nabla^2 Q_\beta = \frac{K k_1 c^2}{R} \left[ \frac{2 k_2 \frac{\partial^2 v}{\partial \alpha^2} - \frac{\partial^3 w}{\partial \beta^3}}{\frac{\partial w}{\partial \beta} - \left( 2 \frac{k_2}{k_1} + \mu_2 \right) \frac{\partial^3 w}{\partial \alpha^2 \partial \beta}} \right] \quad (21)$$

where  $K = E_1 h / (1 - \mu_1 \mu_2)$ ,  $k_1 = E_2 / E_1$ ,  $k_2 = G(1 - \mu_1 \mu_2) / E_1$ ,  $c^2 = h_0^3 / (12 R^2 h)$ .

Using Eqs. (7) and (8), we get Kelvin-like orthotropic elastic shell model.

The obtained model is as follows

$$\begin{aligned} & \left[ \frac{\partial^2}{\partial \alpha^2} + k_2(1 + c^2) \frac{\partial^2}{\partial \beta^2} \right] u + \left[ (\mu_1 + k_2) \frac{\partial^2}{\partial \alpha \partial \beta} \right] v \\ & + \left[ 6 + \frac{\partial}{\partial \alpha} + c^2 \left( k_2 \frac{\partial^3}{\partial \alpha \partial \beta^2} - \frac{\partial^3}{\partial \alpha^3} \right) \right] w \\ & = \frac{\rho h R^2 [1 - \nabla^2] \partial^2 u}{K \partial t^2} \end{aligned} \quad (22)$$

$$\begin{aligned} & \left[ (\mu_1 + k_2) \frac{\partial^2}{\partial \alpha \partial \beta} \right] u + \left[ k_2(1 + 3c^2) \frac{\partial^2}{\partial \alpha^2} + k_1 \frac{\partial^2}{\partial \beta^2} \right] v \\ & + \left[ k_1 \frac{\partial}{\partial \beta} - c^2(\mu_1 + 3k_2) \frac{\partial^3}{\partial \alpha^2 \partial \beta} \right] w \\ & = \frac{\rho h R^2 [1 - \nabla^2] \partial^2 v}{K \partial t^2} \end{aligned} \quad (23)$$

$$\begin{aligned} & \left[ \mu_1 \frac{\partial}{\partial \alpha} - c^2 \left( \frac{\partial^3}{\partial \alpha^3} - k_2 \frac{\partial^3}{\partial \alpha \partial \beta^2} \right) \right] u \\ & + \left[ k_1 \frac{\partial}{\partial \beta} - c^2 (\mu_1 + 3k_2) \frac{\partial^3}{\partial \alpha^2 \partial \beta} \right] v \\ & + \left[ \left( 1 + \frac{1}{c^2} \right) k_1 + \frac{\partial^4}{\partial \alpha^4} + k_1 \frac{\partial^4}{\partial \beta^4} + 2k_1 \frac{\partial^2}{\partial \beta^2} \right. \\ & \left. + (2\mu_1 + 4k_2) \frac{\partial^4}{\partial \alpha^2 \partial \beta^2} \right] c^2 w + \frac{R^2}{K} (1 - \nabla^2) \left[ Ew + \eta \frac{\partial w}{\partial t} \right] \\ & = - \frac{\rho h R^2 [1 - \nabla^2] \partial^2 w}{K \partial t^2} \end{aligned} \tag{24}$$

where  $K = \frac{E_1 h}{1 - \mu_1 \mu_2}$ , medium has stiffness  $E$ , and the viscosity of the medium is  $\eta$ . Two kinds of boundary conditions may be assumed while solving such problems. These three conditions are:

Clamped-clamped

$$\begin{aligned} \alpha = \beta = \gamma = \frac{\partial \gamma}{\partial \alpha} = 0, \\ \text{at } \alpha = 0, \alpha = L/R \end{aligned} \tag{25}$$

Clamped-free

$$\begin{cases} \alpha = \beta = \gamma = \frac{\partial \gamma}{\partial \alpha} = 0 & \text{at } \alpha = 0 \\ N_{\alpha\alpha} = M_{\alpha\alpha} = N_{\alpha\beta} = M_{\alpha\beta} = 0 & \text{at } \alpha = L/R \end{cases} \tag{26}$$

where  $L$  is the length of CNTs.

Using any combination of above three conditions we come close to nonlocal Flugge’s shell model. Above system of equations is the nonlocal orthotropic Kelvin-like shell model for CNTs. To understand the waves propagating in CNTs, we need to derive the dispersion relations.

### 2.2 Application of wave propagation approach

The solutions of wave system of Eqs. (22)-(24) for axisymmetric waves is given by (Wang and Gao 2016).

$$\begin{cases} u(\alpha, t) = U e^{ik(\alpha - \frac{vt}{R})} \\ v(\alpha, t) = V e^{ik(\alpha - \frac{vt}{R})} \\ w(\alpha, t) = W e^{ik(\alpha - \frac{vt}{R})} \end{cases} \tag{27}$$

where  $U, V$  and  $W$  are the amplitudes of waves along the direction of  $x, y$  and  $z$  respectively, the dimensionless wave vector in the longitudinal direction is  $k = \frac{\pi m R}{L}$ , in longitudinal direction  $m, v$  is the half axial wave number and wave phase velocity.

Substituting Eq. (27) in system of Eqs. (22)-(24) and simplifying, in matrix form, we get the following system

$$[M^{(1)}(k, v)]_{3 \times 3} \begin{bmatrix} U \\ V \\ W \end{bmatrix} = [0 \ 0 \ 0]^T \tag{28}$$

For the nontrivial solution of above equation, we have

$$Det[M^{(1)}(k, v)] = 0 \tag{29}$$

Table 1 Convergence of present Kelvin’s model frequencies of SWCNTs with Ref. (Heydarpour *et al.* 2014)

$m$	$\nu$	Kelvin’s theory	Heydarpour <i>et al.</i> (2014)	Present
		7	0.6240	0.6228
	0.12	9	0.6240	0.6234
		11	0.6240	0.6239
0	0.17	7	0.8157	0.8143
		9	0.8157	0.8152
		11	0.8157	0.8155
	0.28	7	0.8553	0.8541
		9	0.8553	0.8547
		11	0.8553	0.8550

Table 2 Convergence of present Kelvin’s model frequencies SWCNTs frequencies for parameters  $\Delta = \omega R \sqrt{(1 - \nu^2) \rho / E}$  with Swaddiwudhipong *et al.* (1995)

$L/R$	$m$	Method	$\parallel$			
			1	2	3	4
20	1	Swaddiwudhipong <i>et al.</i> (1995)	0.016101	0.00545	0.00504	0.008534
		Present	0.016103	0.00545	0.00505	0.008536
0.25	1	Swaddiwudhipong <i>et al.</i> (1995)	0.95193	0.93446	0.90673	0.87076
		Present	0.95194	0.93446	0.90674	0.87078

### 3. Results and discussion

A methodology for constructing the modified Kelvin’s theory based on wave propagation approach of single-walled CNTs for vibration analysis has been presented with the prescribed boundary conditions as: C-C and C-F. For the accuracy of present model, the frequencies of SWCNTs are well matched with the earlier published results. Since the percentage of error is negligible, the model is concluded as valid. The following material parameters are as, the ratio of Young’s modulus and mass density, the in-plane stiffness or rigidity, Poisson’s ratio  $E / \rho = 3.6481 \times 10^8 \text{ m}^2 / \text{s}^2$ ,  $Eh = 278.25 \text{ GPa}\cdot\text{nm}$ ,  $\nu = 0.2$ , respectively (Zhang *et al.* 2009). The presented results are compared with Heydarpour *et al.* (2014) and Swaddiwudhipong *et al.* (1995) in Tables 1, 2 and for the validation with the experimental results (Jorio *et al.* 2001) as shown in Table 3. The percentage error of present results is in the acceptable range with  $\pm 5\%$ . We can observe the present results of Kelvin’s model having good coherence with Heydarpour *et al.* (2014), Swaddiwudhipong *et al.* (1995) and Jorio *et al.* (2001).

Now the results for armchair, zigzag and chiral are discussed in detail with the present Kelvins model. For armchair frequencies computations of C-C and C-F BCs is obtained with varying index 10~20 and  $L/d = 4.86\sim 35.53$ . The fundamental natural frequencies (THz) of presented BCs are noted as 0.97293 to 0.011812 and 0.76587 to 0.01640, respectively, as shown in Figs. 4-6. Now the study

Table 3 Convergence of present Kelvin’s model with experimental model of Jorio *et al.* (2001)

$(m, n)$	$f$ (THz)							
	(8, 7)	(10, 5)	(11, 4)	(14, 1)	(18, 0)	(17, 2)	(11,11)	(12, 12)
Jorio <i>et al.</i> (2001)	7.165	7.105	6.865	6.295	5.276	5.216	4.917	4.527
Present	7.018	7.093	5.781	6.103	5.161	5.142	4.805	4.465
Difference %	1.42	1.49	3.85	1.92	1.76	1.83	1.87	1.71

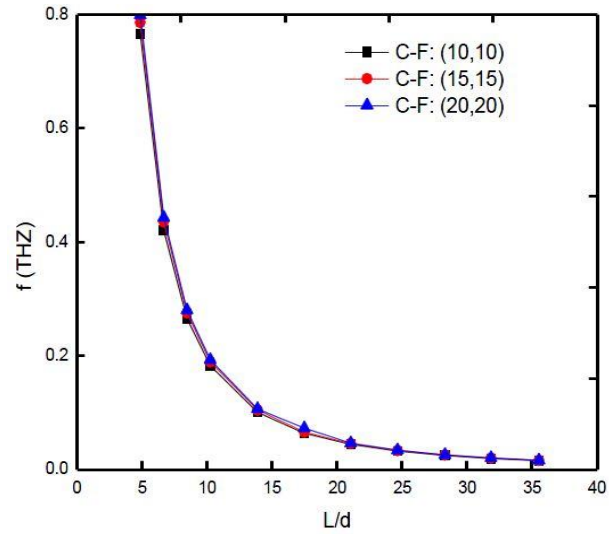
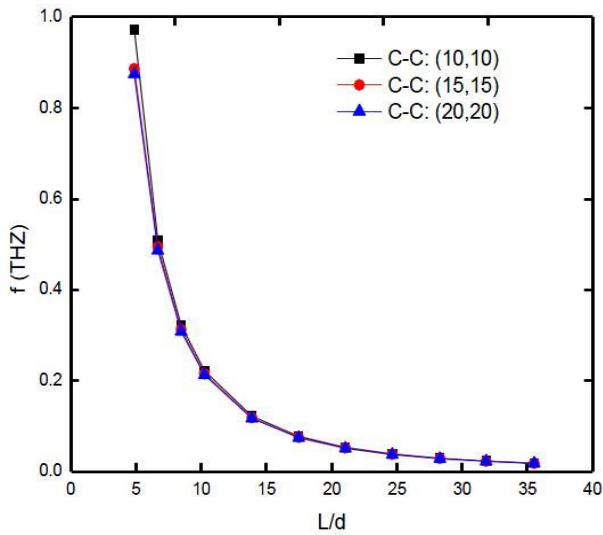


Fig. 4 Frequencies of armchair (10, 10), (15, 15), (20, 20) SWCNTs versus  $L/d$

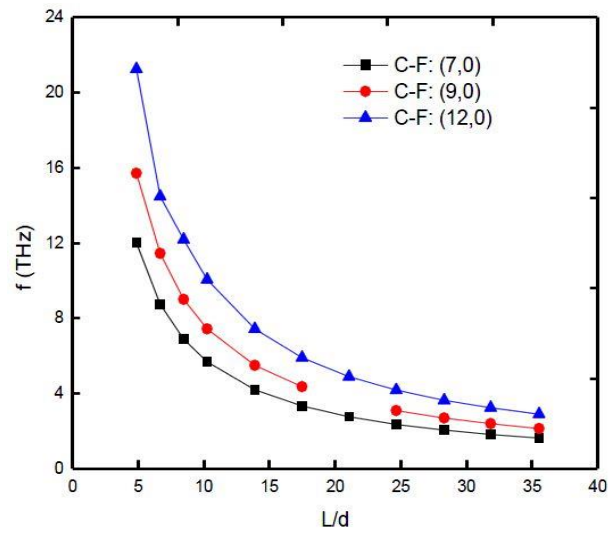
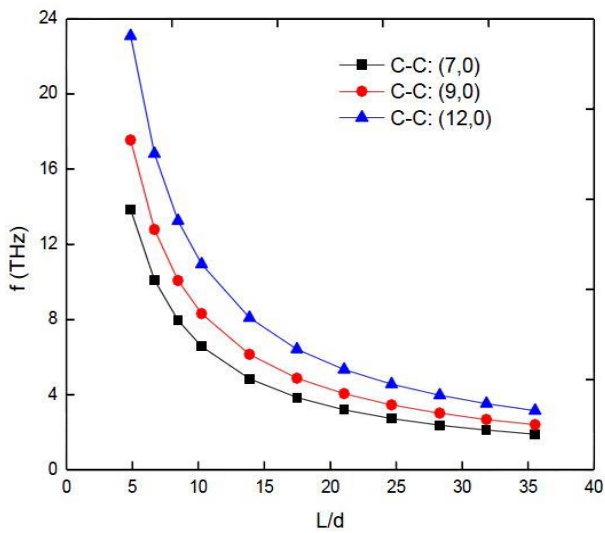


Fig. 5 Frequencies of zigzag (7, 0), (9, 0), (12, 0), SWCNTs versus  $L/d$

is shifted to six zigzag tube measurement based on wave propagation approach by changing index 7~12, keeping ratios  $L/d$  fixed with BCs. The data set points of C-C (C-F) are obtained from 13.855~3.1585 THz (2.007~2.9058 THz) using present approach as shown in the Fig. 5. In all of these modes, (7, 0), the cross sections do not deform significantly from the nearly circular perimeter. So, for the other zigzag case (9, 0) and (12, 0), the natural frequencies increase, the axial and circumferential directions develop and become more visible. For chiral SWCNTs, with same

length-to-diameter ratio, varying the chiral index as (6, 4), (9, 3), and (16, 4), it is observed that the frequency curve for chiral index (16, 7) is composed between (6, 4) and (9, 3). The computations for index (6, 4), are found to be 1.853 to 0.044 0.97293 to 0.011812 THz (1.413 to 0.032 THz), for index (9, 3) as 4.369 to 0.161 THz (3.875 to 0.130THz) and for index (16, 7) as 3.222 to 0.094 THz (2.970 to 0.083 THz), respectively, as shown in Fig. 6. Figs. 7-8 contain the frequencies in THz for in-plane rigidity of SWCNTs against the ratio of length-diameter  $L/d$ . These figures exhibit that

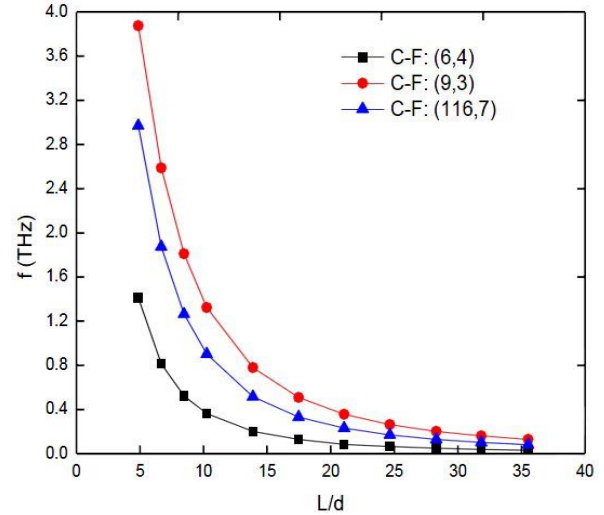
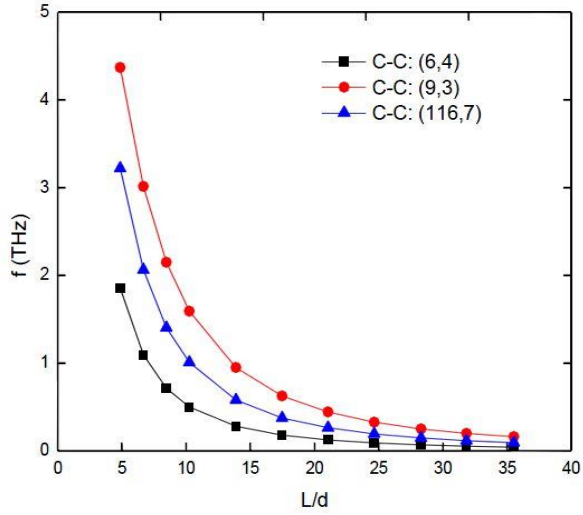
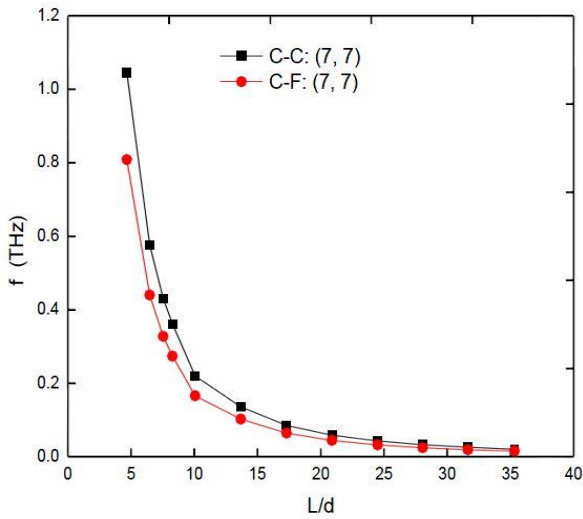
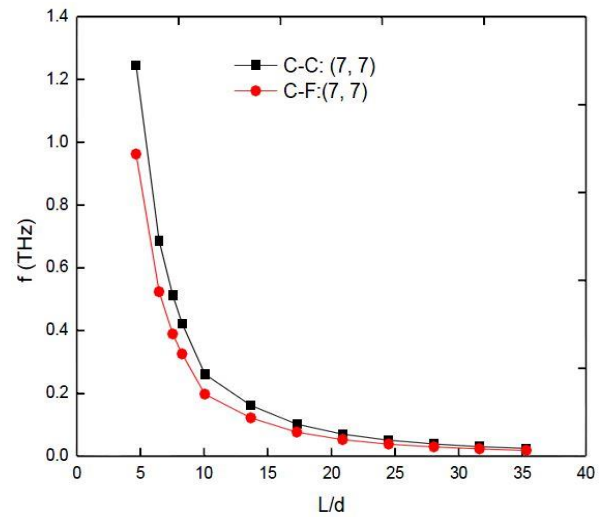


Fig. 6 Frequencies of chiral (6, 4), (9, 3), (16, 7), SWCNTs versus  $L/d$

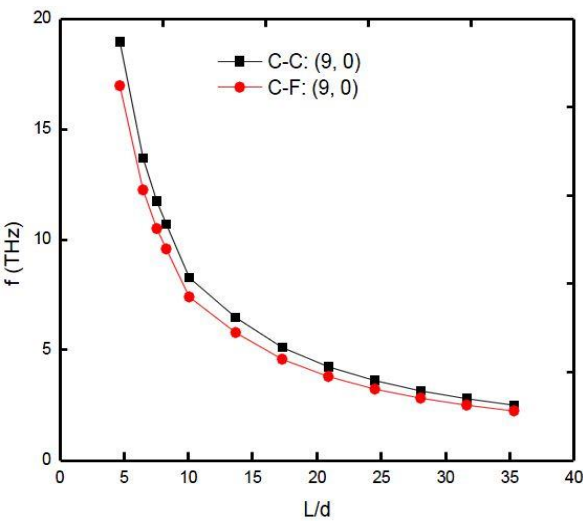


(a)

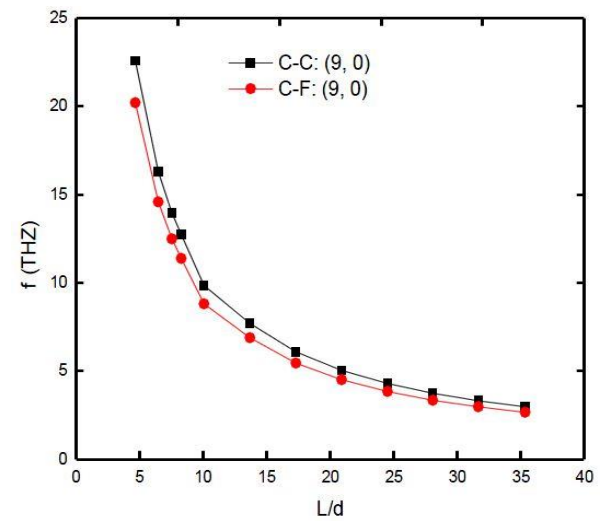


(b)

Fig. 7 Frequencies of armchair (7, 7) SWCNTs versus  $L/d$  with  $Eh$  (a) 300 GPa.nm; (b) 425 GPa.nm



(a)



(b)

Fig. 8 Frequencies of zigzag (9, 0) SWCNTs versus  $L/d$  with  $Eh$  (a) 300 GPa.nm; (b) 425 GPa.nm



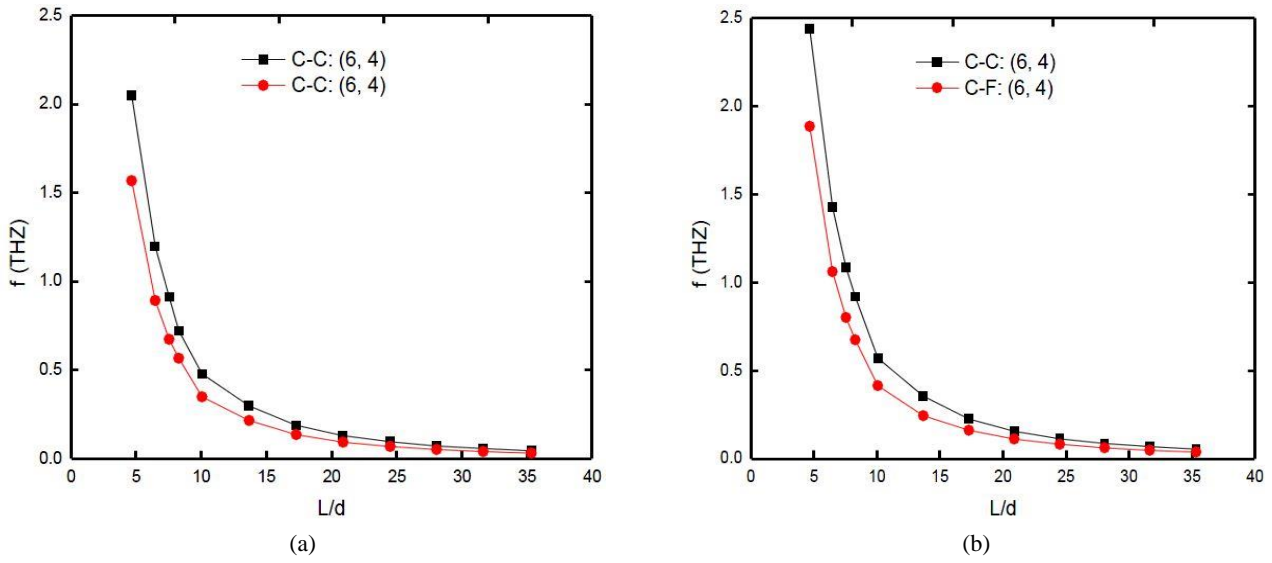


Fig. 9 Frequencies of chiral (6, 4) SWCNTs versus  $L/d$  with  $Eh$  (a) 300 GPa.nm; (b) 425 GPa.nm

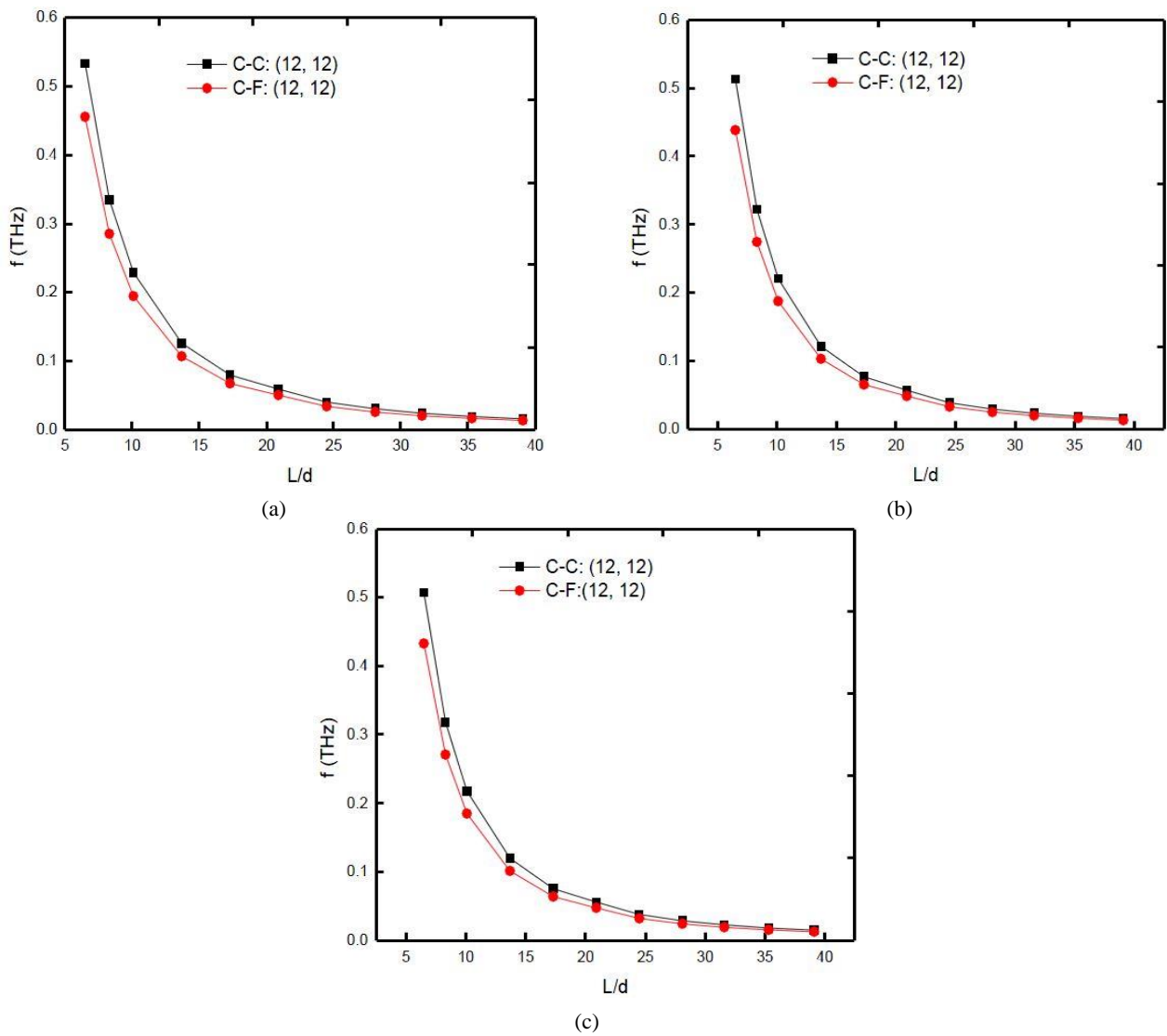


Fig. 10 Frequencies of zigzag (12, 12) SWCNTs versus  $L/d$  with  $ph$  (a) 740.52 nm; (b) 800.64 nm; (c) 820.80 nm

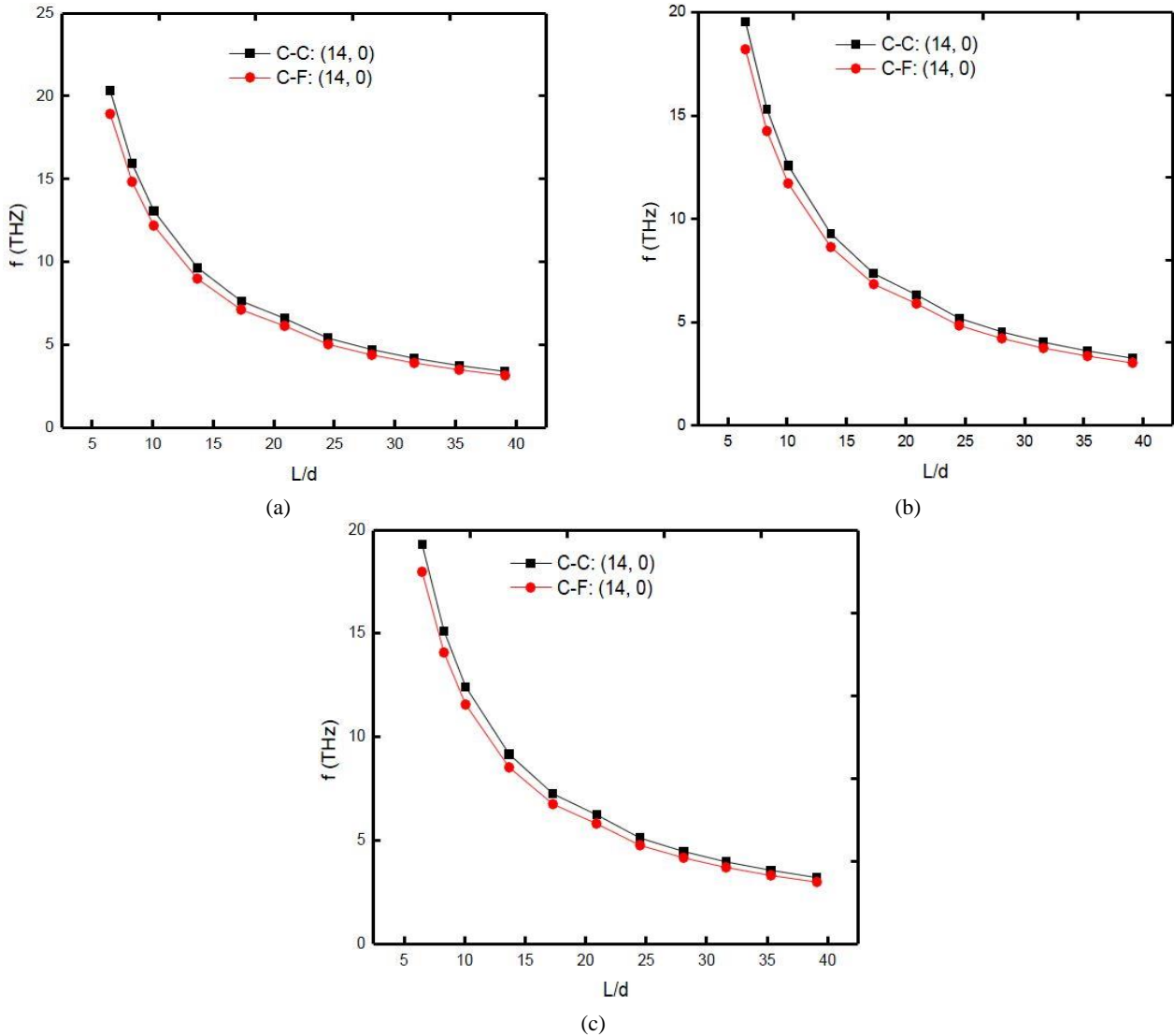


Fig. 11 Frequencies of zigzag (14, 0) SWCNTs versus  $L/d$  with (a) 740.52 nm; (b) 800.64 nm; (c) 820.80 nm

in-plan rigidity  $Eh$  varying from 300 GPa·nm ~425. Natural frequencies of SWCNTs for the category armchair (7, 7) and zigzag (9, 0) with specified boundary conditions using WPA are plotted against the length-diameter ratio, respectively. An interesting phenomenon can be seen from Figs. 6-7, that the fundamental natural frequency of zigzag (9, 0) SWCNTs are substantially larger than that of armchair (7, 7) SWCNTs. It is mentioned that the frequencies with boundary condition increase with an increase in IPR ( $Eh$ ) and fundamental frequency decrease as  $L/d$  increase. The main result of the fundamental natural frequency (THz) at  $Eh = 300$  GPa·nm for armchair (7, 7) as C-C (C-F) = 1.04524 (0.80828) increases to C-C (C-F) = 1.24408 (0.9620) at  $Eh = 425$  GPa·nm,  $L/d = 4.67$  as shown in Fig. 7 and at  $Eh = 425$  GPa·nm for zigzag (9, 0) as C-C (C-F) = 18.963 (16.967) increases to C-C (C-F) = 22.570 (20.185) at  $Eh = 425$  GPa·nm keeping  $L/d$  fixed as shown in Fig. 7.

Furthermore, the main result of the frequency (THz) at  $Eh = 300$  GPa·nm for armchair (7, 7) as C-C (C-F) = 0.020594 (0.01547) slightly changes to 0.02451 (0.01842)

at  $Eh = 425$  GPa·nm and  $L/d = 4.67$  as shown in Fig. 7 and at  $Eh = 425$  GPa·nm for zigzag (9, 0) as C-C (C-F) = 2.5059 (2.2421) increases to C-C (C-F) = 2.9826 (2.6686) at  $Eh = 425$  GPa·nm keeping  $L/d$  fixed as shown in Fig. 8. It is observed that with the slight increase in  $L/d$ , the frequencies for both armchair and zigzag decrease constantly instead for higher values of  $Eh$ . Fig. 9 shows the variation of frequencies with the effect of in-plane rigidity of chiral SWCNTs versus  $L/d$ . Also, in these figures as above discussed in the case of zigzag and chiral, in-plane rigidity  $Eh$  varying from 300 GPa·nm ~425 GPa·nm. The computed frequencies C-C and C-F BCs are presented with proposed model based on FSM. It can be seen that for smaller ratio, the frequencies are more significant and prominent. For example, with the end condition, as C-C (C-F) = (6, 4)  $f \sim 2.048$  (1.568) at  $Eh = 300$  GPa·nm increases to C-C (C-F) = (6, 4)  $f \sim 2.438$  (1.866) at  $Eh = 425$  GPa·nm for chiral SWCNTs as shown in Fig. 9. It is important to note that the fundamental natural frequencies of clamped-clamped for all indices are normally over-predicting than clamped-free. These figures show the aspect ratios with frequencies for in-

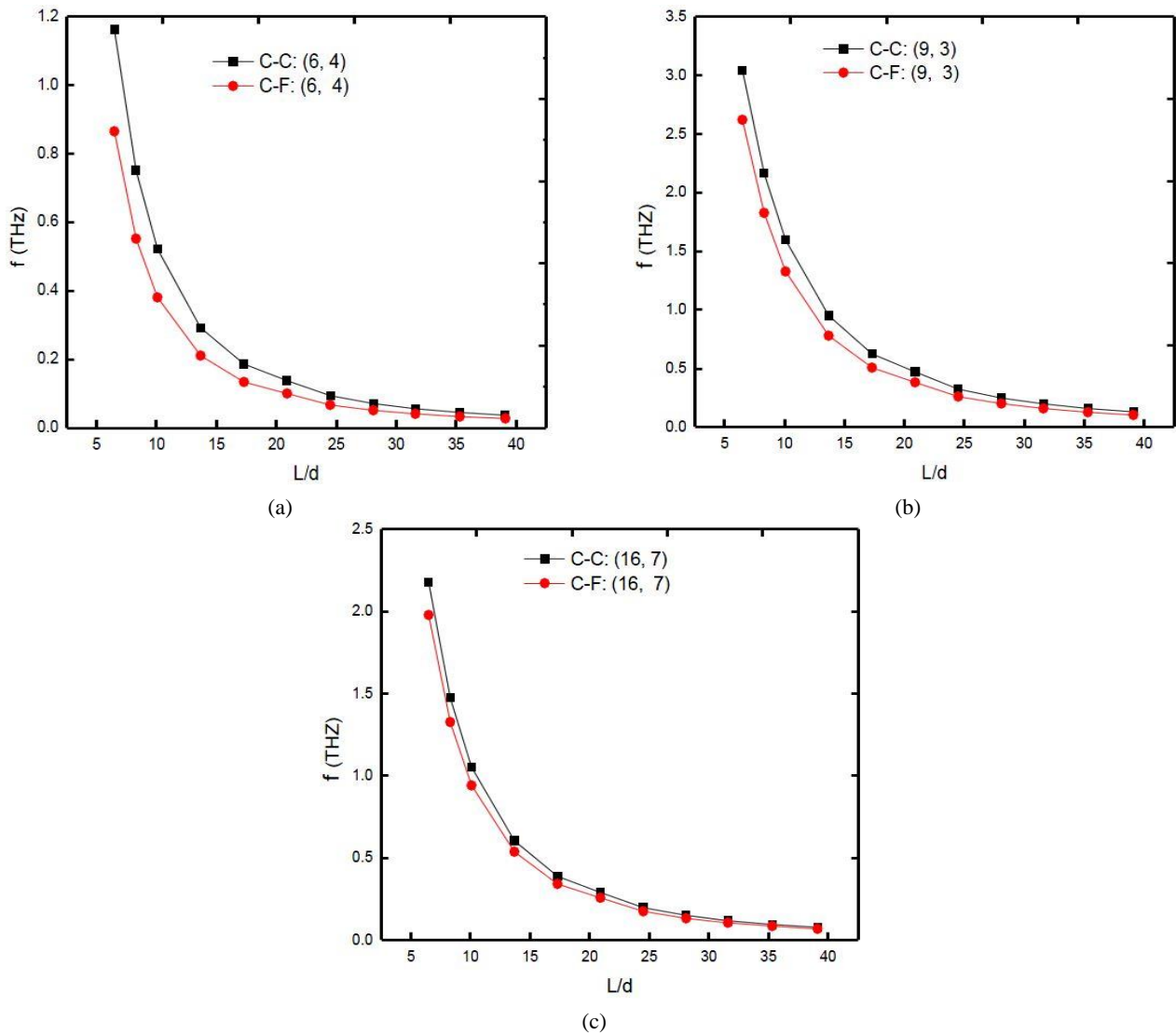


Fig. 12 Frequencies of chiral (6, 4), (9, 3), (16, 7) SWCNTs versus  $L/d$  with  $\rho h = 740.52$  nm

plane rigidity  $Eh = 300\text{--}425$  GPa.nm. A trend of increasing frequencies of indices is as  $(9, 3) > (16, 7) > (6, 4)$ . The chiral frequency decreases constantly instead for higher values of  $Eh$  with the slight increase in  $L/d$ , the frequencies for Figs. 10-11 exhibit the variation of C-C and C-F frequency for the case of the MDA ( $\rho h$ ). Results show that due to variation of MDA on frequency decreases considerably where the phenomena remain same for the in-plane rigidity case with same end conditions. In these figures, variation of armchair (12, 12) and zigzag (14, 0) is plotted for different values of  $\rho h = 740.52$  nm, 800.64 nm and 820.80 nm under prescribed boundary conditions. The fundamental natural frequencies for armchair tubes at  $\rho h = 740.52$  nm, C-C (C-F) = (12, 12)  $f \sim 0.53352$  (0.45575) decreases to C-C (C-F) = (12, 12)  $f \sim 0.50676$  (0.43289) at  $\rho h = 820.80$  nm and also at  $\rho h = 820.64$  nm decreases to C-C (C-F) = (12, 12)  $f \sim 0.50676$  (0.43289) for  $L/d = 6.5$ , as shown in Fig. 9 and for zigzag SWCNTs at  $\rho h = 740.52$  nm C-C (C-F) = (14, 0)  $f$  (THz)  $\sim 20.325$  (18.923) decreases at  $\rho h = 820.80$  nm C-C (C-F) = (14, 0)  $f \sim 19.3057$  (17.974) and at  $\rho h = 820.64$  nm, C-C (C-F) = (14, 0)  $f \sim 19.3057$  (17.974)

and  $L/d = 6.5$  as shown in Fig. 10. It is observed that for all the cases, with increasing in  $\rho h$  the resulting fundamental natural frequencies decrease.

Figs. 12-14 show the length-to-diameter ratios with frequencies for different values of  $\rho h = 740.52$  nm, 800.64 nm and 820.80 nm. It is noted that, with the increasing of ratio of length/diameter and  $\rho h$ , the frequencies decreases of the CNTs as C-C (C-F) = (6, 4)  $f \sim 1.162$  (0.865), C-C (C-F) = (9, 3)  $f \sim 3.162$  (2.723), C-C (C-F) = (16, 7)  $f \sim 2.178$  (1.979) at  $\rho h = 740.52$  nm decreases to C-C (C-F) = (6, 4)  $f \sim 1.118$  (0.832), C-C (C-F) = (9, 3)  $f \sim 1.103$  (0.821), C-C (C-F) = (16, 7)  $f \sim 3.003$  (2.586) at  $\rho h = 800.64$  nm for decreases to C-C (C-F) = (6, 4)  $f \sim 3.043$  (2.621), C-C (C-F) = (9, 3)  $f \sim 2.096$  (1.9048), C-C (C-F) = (16, 7)  $f \sim 2.068$  (1.879) at  $\rho h = 820.80$  nm for chiral SWCNTs and  $L/d = 6.5$  as shown in Figs. 10-12. In these figures, a reasonable decline in curve suggests that frequencies decrease with increase in MDA  $\rho h = 740.52$  nm to 820.80 nm. From our results, one can easily conclude that the declined in frequencies for MDA curves (6, 4), (9, 3) & (16, 7) is as follows  $(6, 4) < (16, 7) < (9, 3)$ . Interestingly, in all cases C-F frequency curves are

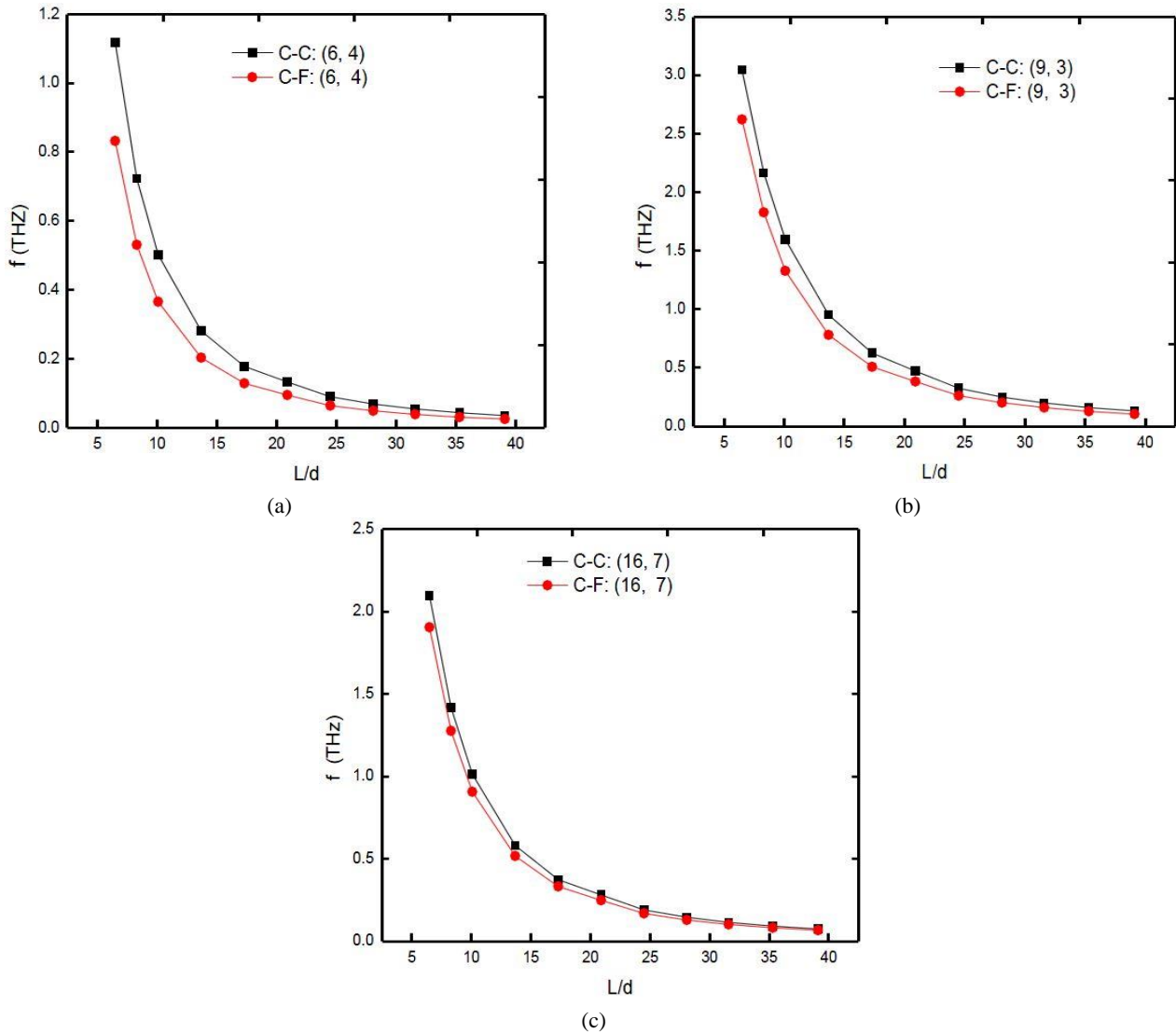


Fig. 13 Frequencies of chiral (6, 4), (9, 3), (16, 7) SWCNTs versus  $L/d$  with  $\rho h = 800.64$  nm

lower than C-C, but follow the same pattern as discussed earlier. Also, the frequency for shorter tube ratio is more significant.

#### 4. Conclusions

In this paper attempts have been made to find the vibrational behavior of SWCNTs based on Kelvin's model by using wave propagation approach under C-C and C-F boundary conditions. It is concluded that from this type of analysis that CNTs behave like shell structure. The fundamental natural frequency of SWCNTs versus ratio of length-to-diameter for a wide range has been reported and investigated through the study with specified boundary conditions. Armchair, zigzag and chiral structures are considered for the vibrational analysis to investigate the effect of different mode and in-plane rigidity with end conditions. Moreover, the influence of MDA on the same structure is also developed. It is also concluded that the fundamental frequencies of zigzag and chiral tubes are

greater than that of armchair tubes frequencies. In addition, by increasing three different value of IPR resulting frequencies also increase and frequencies decrease on increasing mass density per unit lateral area. Though the trends of frequency values of both MDA and IPR are converse to each other. By considering present model based on the Winkler and Pasternak foundations, the vibration characteristics of SWCNTs can be studied.

#### Declaration of Conflicting Interests

The author(s) declared no potential conflicts of interest with respect to the research, authorship, and/or publication of this article.

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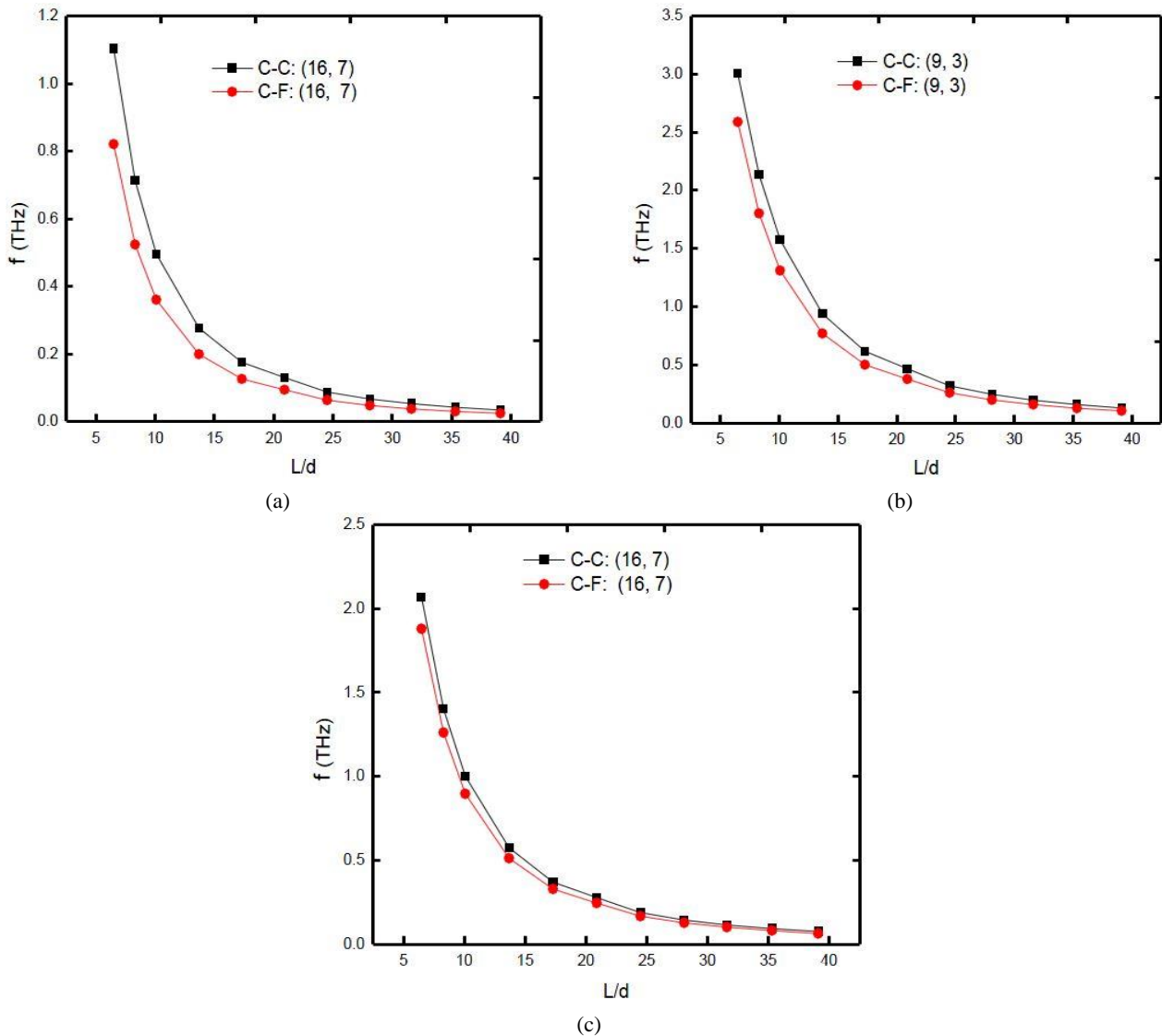


Fig. 14 Frequencies of chiral (6, 4), (9, 3), (16, 7) SWCNTs versus  $L/d$  with  $\rho h = 820.80$  nm

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